

**"Vorticity deposition and evolution
in accelerated inhomogeneous flows:
Analysis, Computation, Experiment & Models"**

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CSCAMM, INC '06

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“AIFS” Accelerated Inhomogeneous Flows

- **Domains:** Supernovae Astrophysics; GFD and Breaking Waves; Laser (IC) Fusion; Supersonic Combustion
- **Objectives:** Understand & model vortex physics & mixing
- **Approach:** Construct Reduced Models via Theory, Simulation, and *Visiometrics* with Experimental Juxtaposition
- **Specific Configurations:** Shock & Forced Acceleration Interactions with various geometries : Perturbed planar & Shock Cylinder

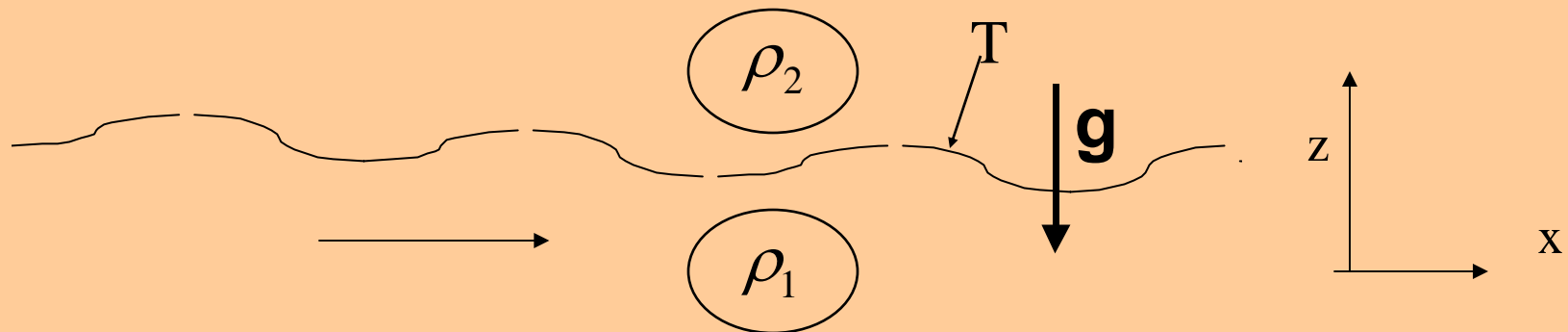
OVERVIEW: “AIFS” 2D Richtmeyer-Meshkov

➤ Topics

- Well-posedness and *finite initial transition layer*
- RM $a\text{-dot}$ \rightarrow *constant* at intermediate times
- Circulation generation (*vortex bilayers*) & gradient Intensification
- Vortex Projectiles & Bounding box elongation
- Baroclinic Turbulence & Forcing

Rayleigh-Taylor & Richtmyer-Meshkov Incompressible

[For RT, see *Hydrodynamic & Hydromagnetic Stability* by S. Chandrasekhar, p.483]



1. Rayleigh-Taylor, $g = \text{constant}$
2. Richtmyer-Meshkov, $g = G \delta(t)$, (impulse)

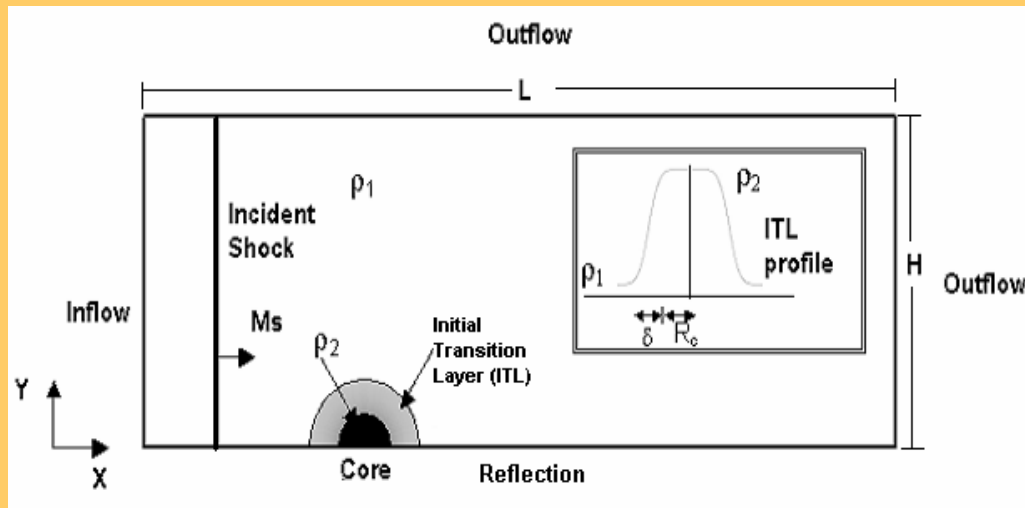
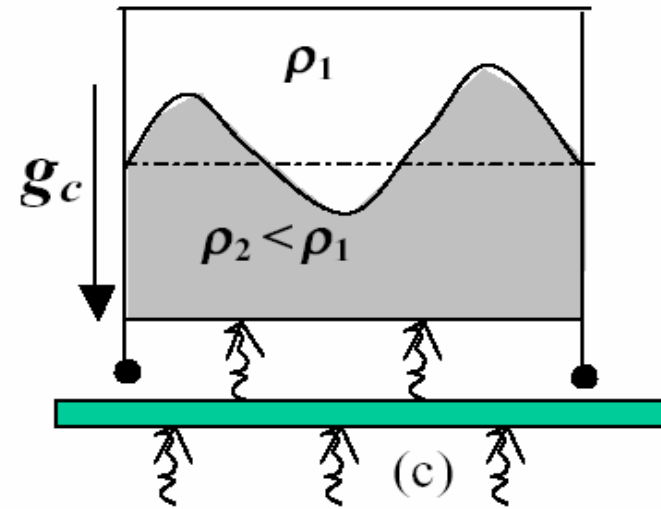
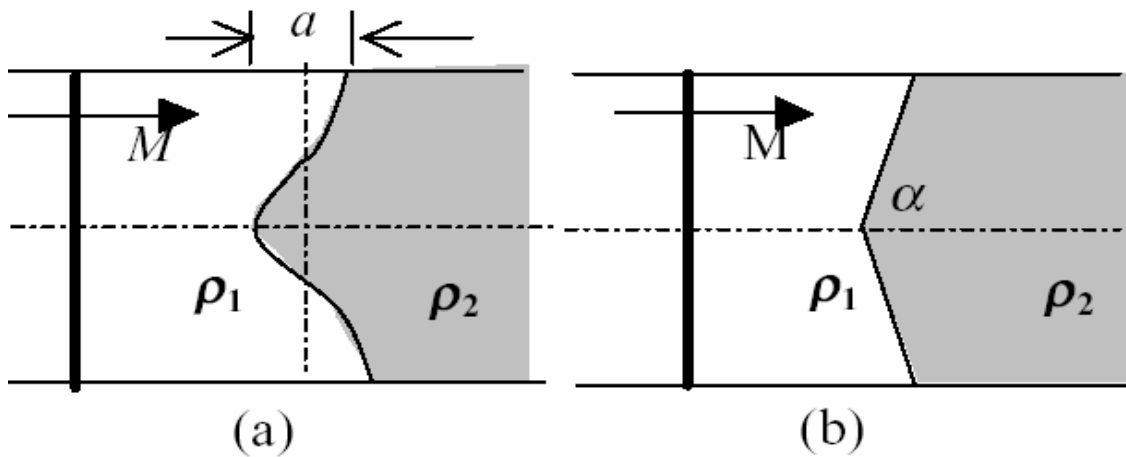
1. Rayleigh-Taylor, $g = \text{constant}$

Taylor's Amplitude Formula

$$\omega = \pm \left[g k_z A + k_z^3 \frac{T}{(\rho_1 + \rho_2)} \right]^{1/2}, \quad \frac{d^2 a(t)}{dt^2} = k g A a(t)$$

where $A = \frac{(\rho_1 - \rho_2)}{(\rho_1 + \rho_2)}$ and g is directed from 2 to 1.

Geometries

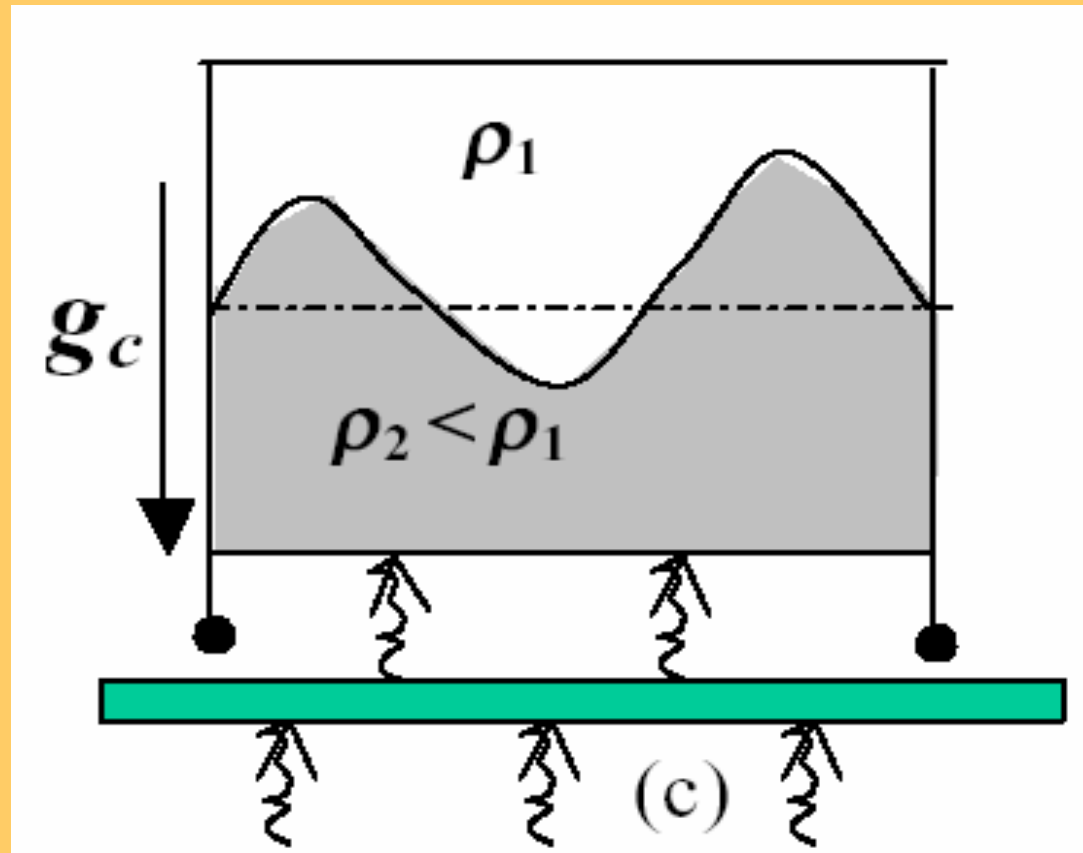


Common geometries in studying *aifs* flows

$$M = \text{Mach No.}, \quad A = \text{Atwood No.} = \frac{\rho_1 - \rho_2}{\rho_1 + \rho_2}$$

Dropped Tank, Incompressible

Jeff Jacobs, Pioneering Experiments

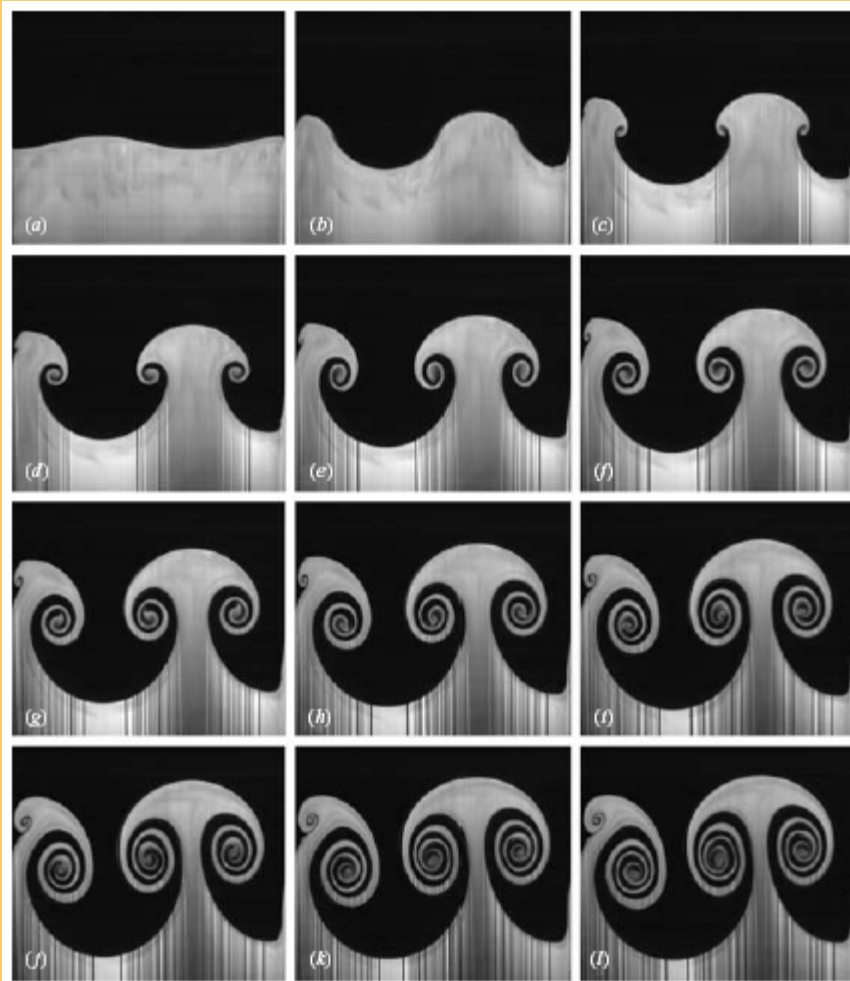


$$M = \text{Mach No.}, \quad A = \text{Atwood No.} = \frac{\rho_1 - \rho_2}{\rho_1 + \rho_2}$$

Jacobs & Niederhaus , Experiment, 2003

Sequence of images from an experiment with 1 & 1/2 waves and $ka_0=0.23$.

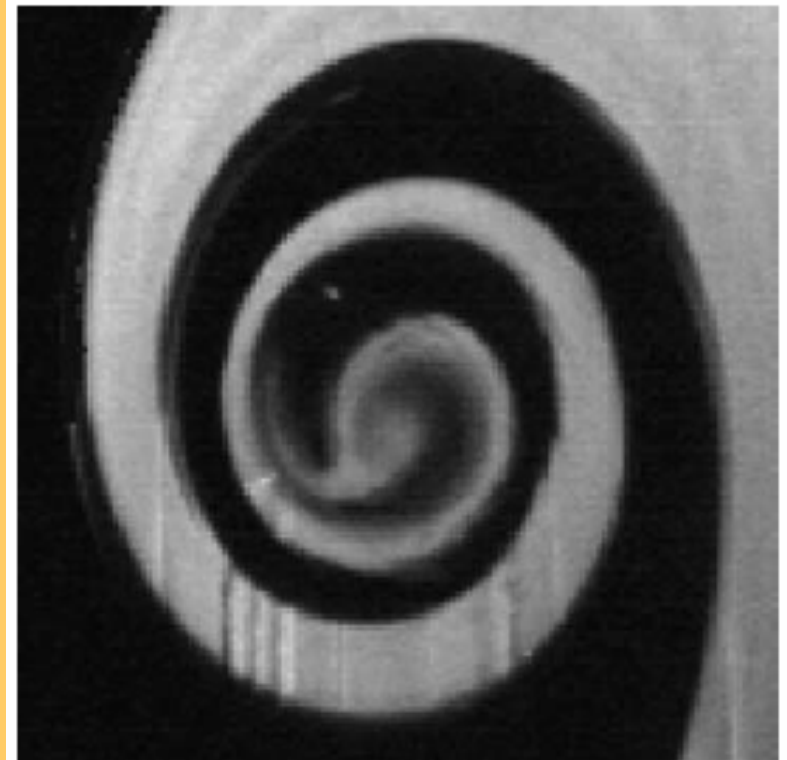
120 mm



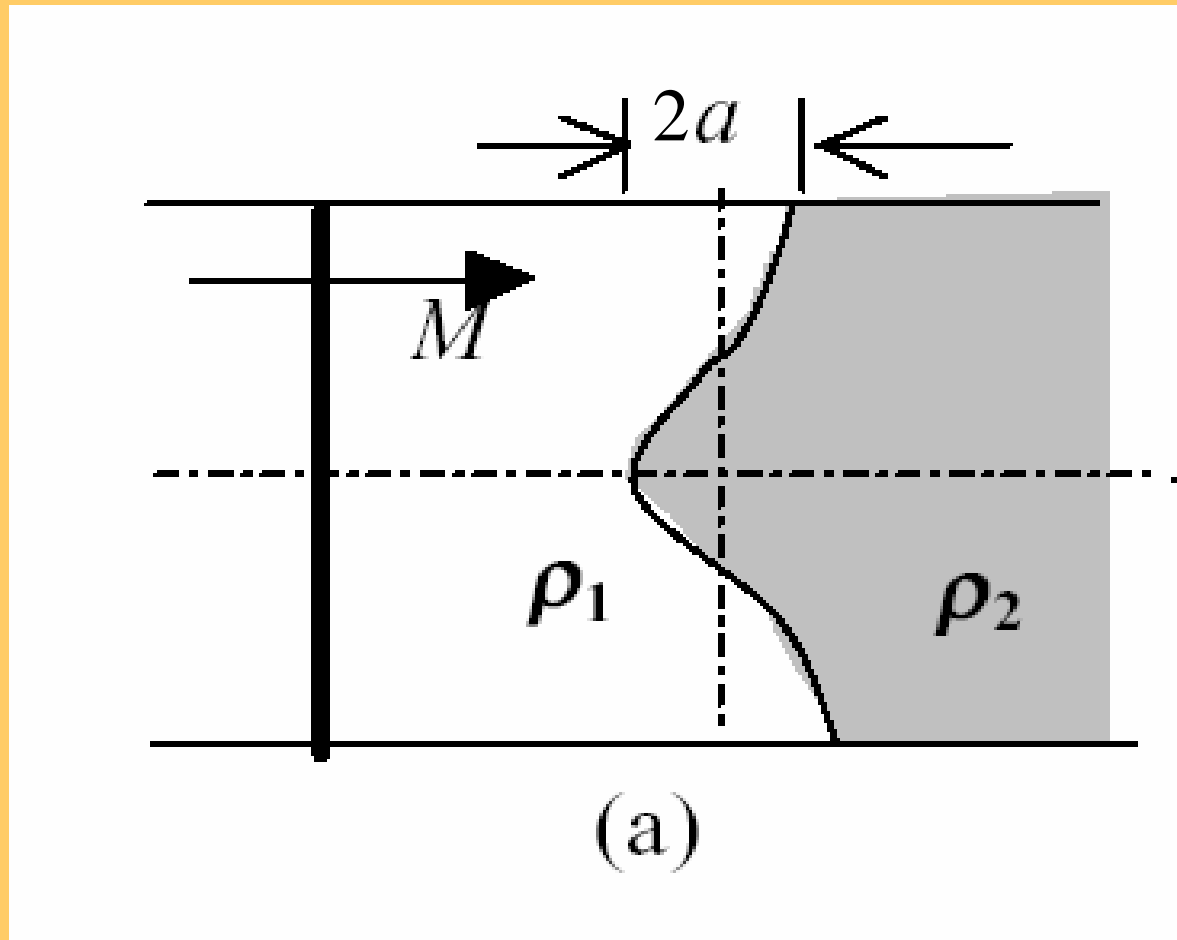
Times relative to the midpoint of spring impact are: (a) -14 ms, (b) 102 ms, (c) 186 ms, (d) 269 ms, (e) 353 ms, (f) 436 ms, (g) 520 ms, (h) 603 ms, (i) 686 ms, (j) 770 ms, (k) 853 ms, (l) 903 ms.

Thick
51mm

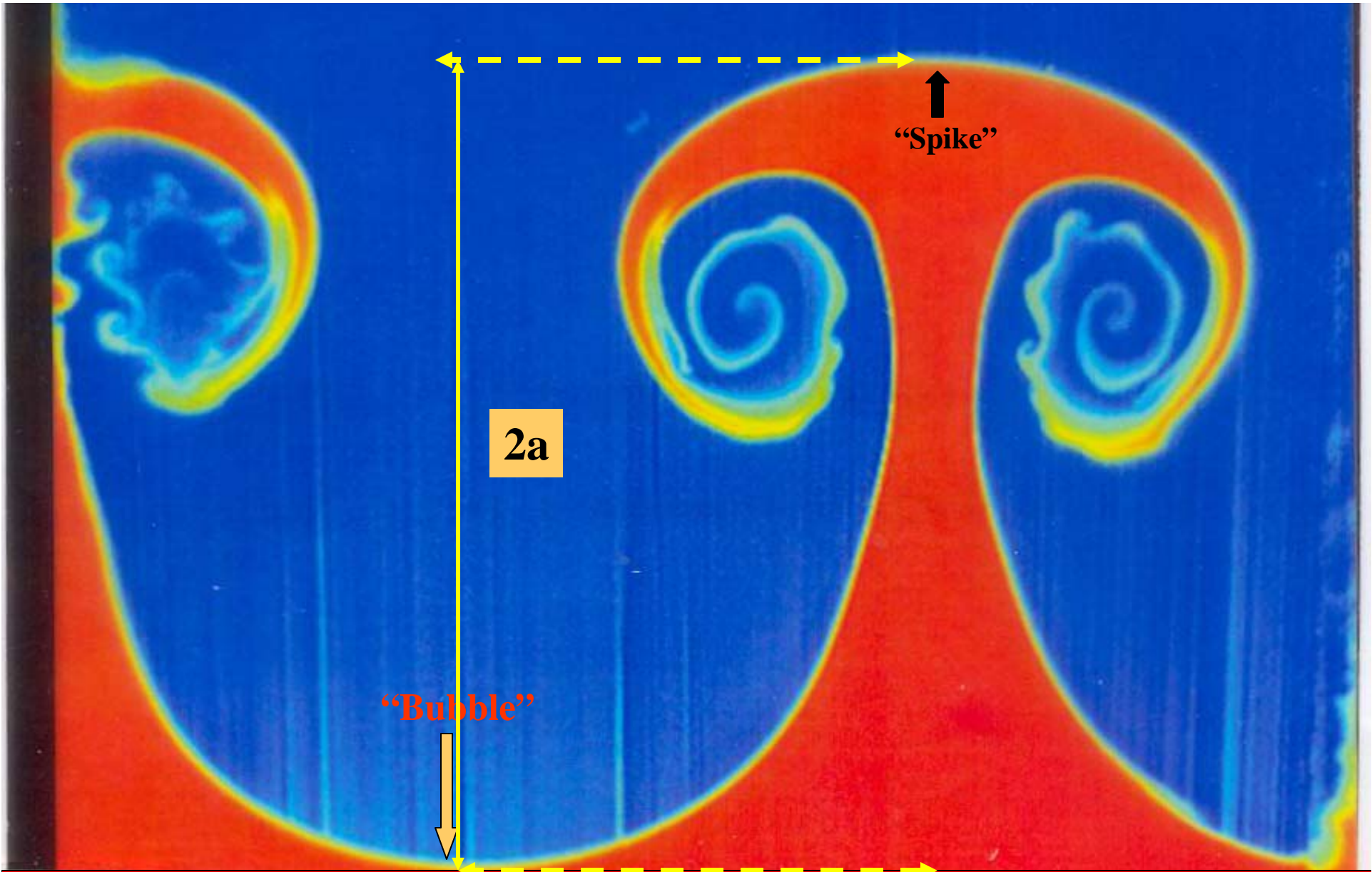
254mm



Classical RM Geometry

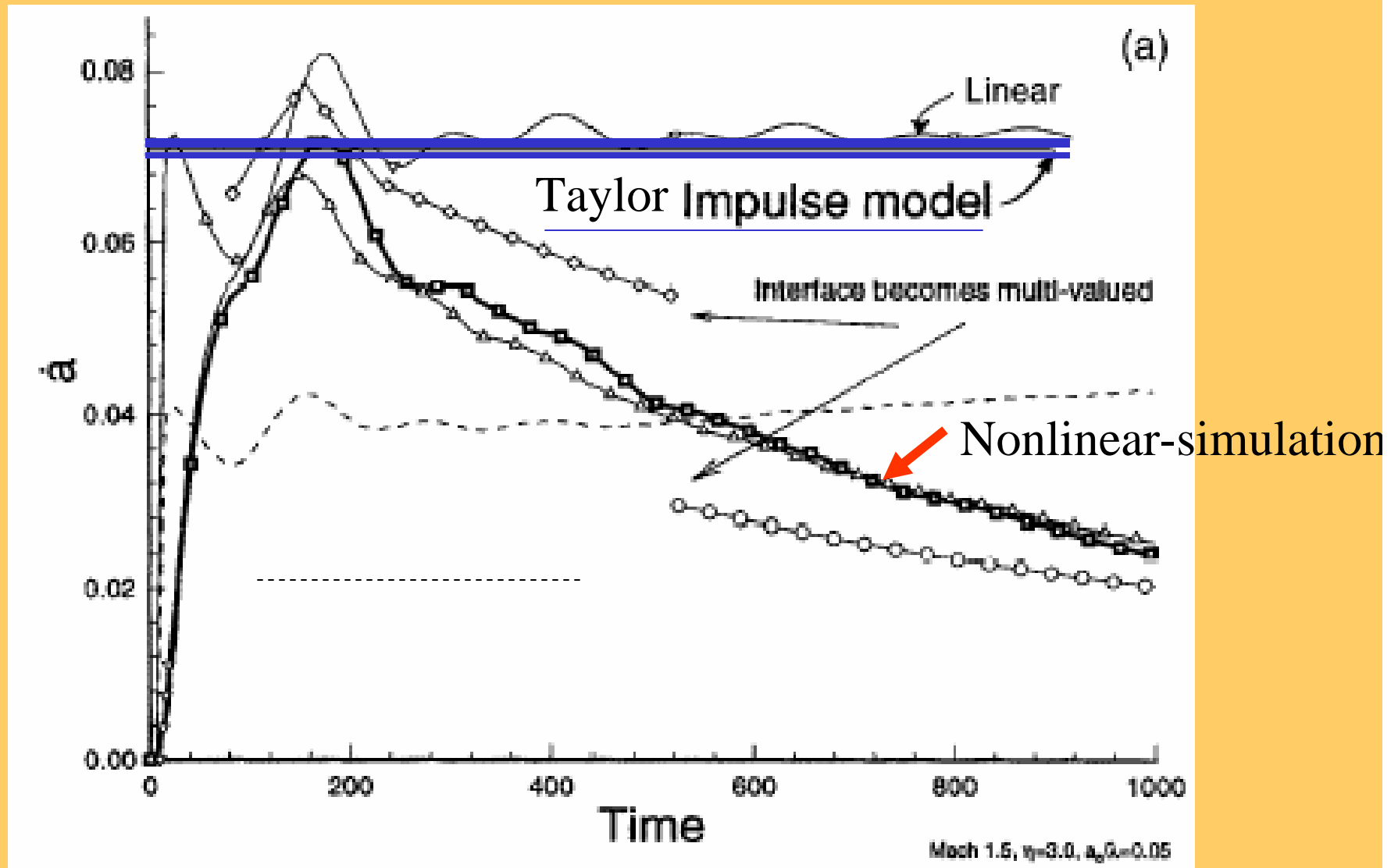


$$M = \text{Mach No.}, \quad A = \text{Atwood No.} = \frac{\rho_1 - \rho_2}{\rho_1 + \rho_2}$$



J. JACOBS & Colleagues, Laboratory Experiments

R-M a-dot comparisons



$$M=1.5, A=0.5, a_0/\lambda=0.05$$

Gas Dynamics Euler Equations for 2D Compressible *RM* Simulations

$$\begin{bmatrix} \rho \\ \rho u \\ \rho v \\ E \end{bmatrix}_t + \begin{bmatrix} \rho u \\ \rho u^2 + p \\ \rho u v \\ u(E + p) \end{bmatrix}_x + \begin{bmatrix} \rho v \\ \rho u v \\ \rho v^2 + p \\ v(E + p) \end{bmatrix}_y = 0$$

where, total energy $E = e + (u^2 + v^2) / 2$

Equations of State (EOS): $p = (\gamma - 1)\rho e$ (for closure)

Vorticity Evolution Equation

$$\frac{\partial \boldsymbol{\omega}}{\partial t} = -\mathbf{u} \cdot \nabla \boldsymbol{\omega} + \boldsymbol{\omega} \cdot \nabla \mathbf{u} - \boldsymbol{\omega}(\nabla \cdot \mathbf{u}) + \frac{1}{\rho^2}(\nabla \rho \times \nabla p)$$

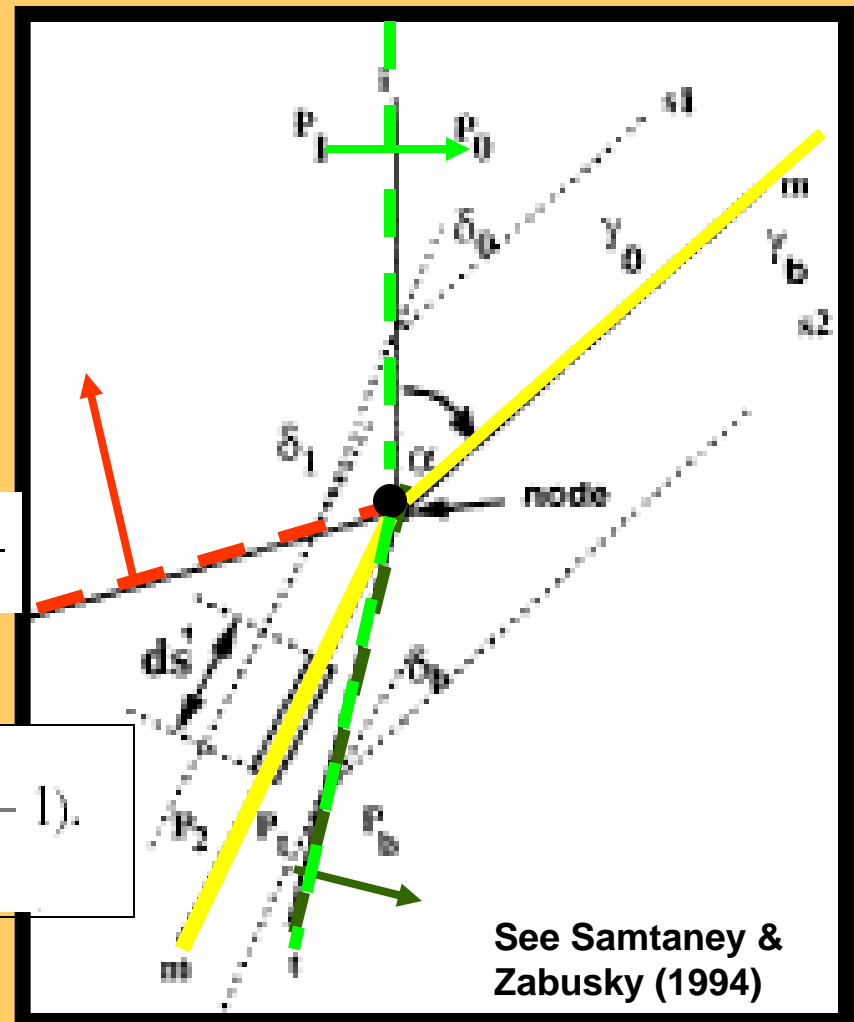
Schematic of regular refraction, for three shocks at a fast-slow interface. incident, i , reflected, r , and transmitted, t , shocks intersecting at a node on the interface, m .

Local, Shock-Polar Analysis Yields Circulation pu Length

$$\frac{d\Gamma}{ds'} \equiv v_t - v_2, \quad \Gamma' \equiv \frac{d\Gamma}{ds} = (v_t - v_2) \frac{\cos\alpha}{\cos(\alpha - \delta_b)}$$

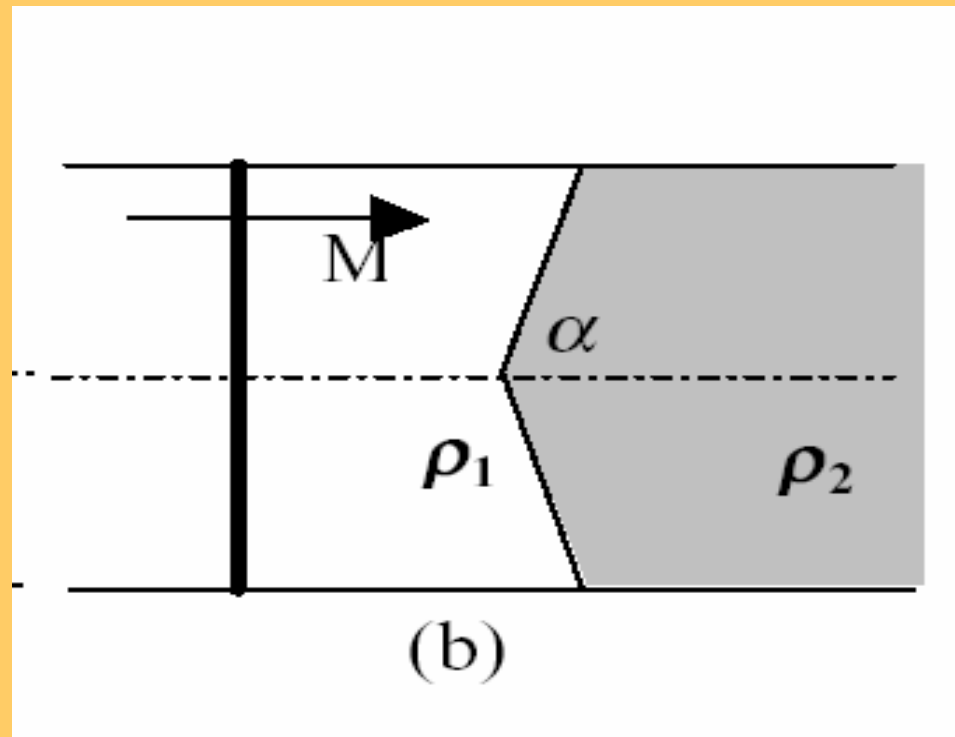
$$\Gamma'_+ = \frac{2\gamma^{\frac{1}{2}}}{\gamma + 1} (1 - \eta^{-\frac{1}{2}}) \sin\alpha (1 + M^{-1} + 2M^{-2})(M - 1).$$

SAMTANEY-ZABUSKY FORMULA



See Samtaney & Zabusky (1994)

Inclined Planar

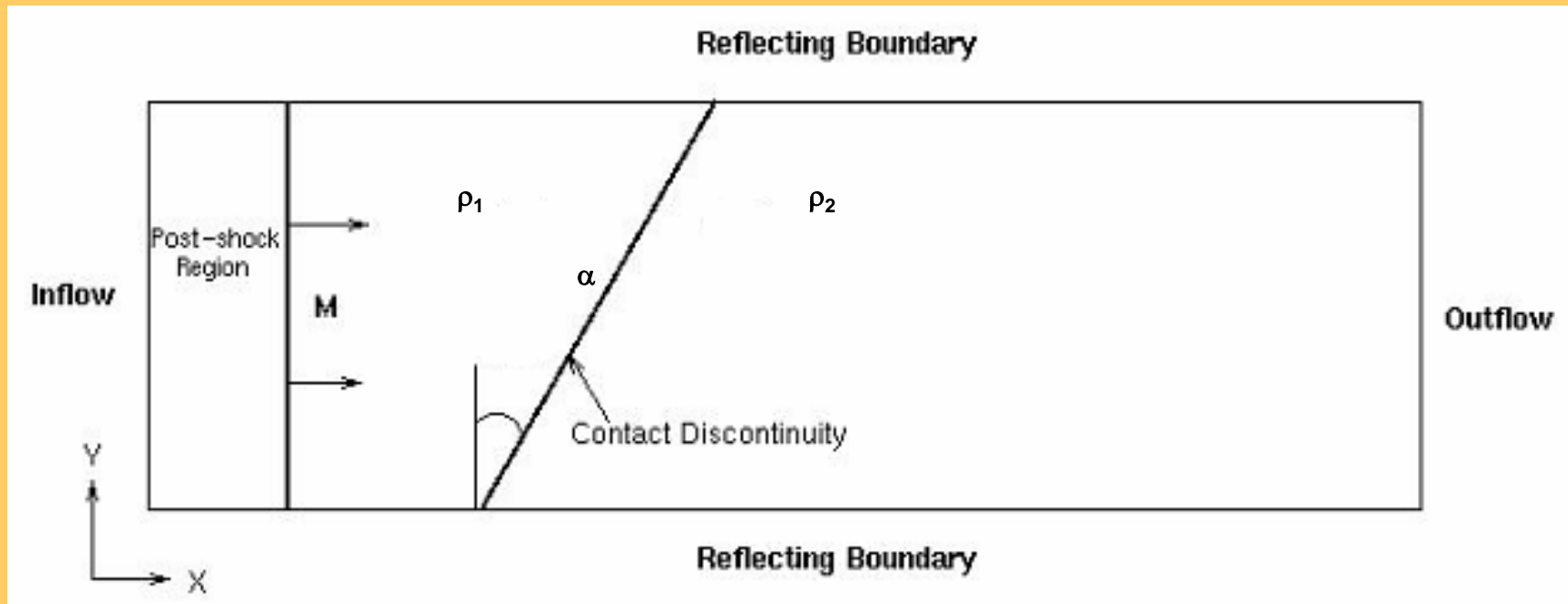


Use symmetry

$$M = \text{Mach No.}, A = \text{Atwood No.} = \frac{\rho_1 - \rho_2}{\rho_1 + \rho_2}$$

Richtmyer-Meshkov Planar inclined Interface B. Sturtevant, pioneering Experiments

Vortex Paradigm, Hawley & Zabusky
PRL, 1989



OVERVIEW: RM

A Vortex Approach

- Topics
 - Well-posedness and
 - *finite initial transition layer (ITL)*

RT & RM Finite-Time singularity

(ill-posed nature of vortex *sheet* evolution)

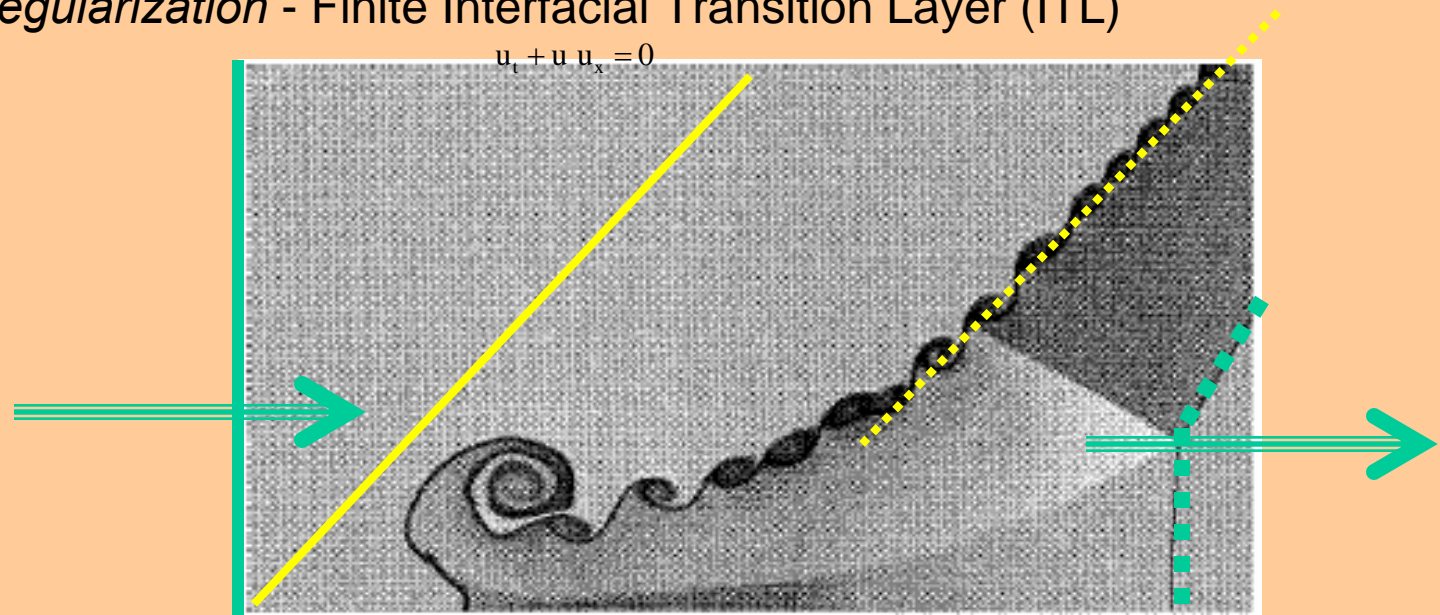
Kelvin-Helmholtz of vortex sheets

– *Finite-time Moore curvature singularity*

- Similar to finite time singularity in first derivative of $u_t + uu_x = 0$

– non-controllable *numerical rollups* due to grid perturbations

– *Regularization* - Finite Interfacial Transition Layer (ITL)



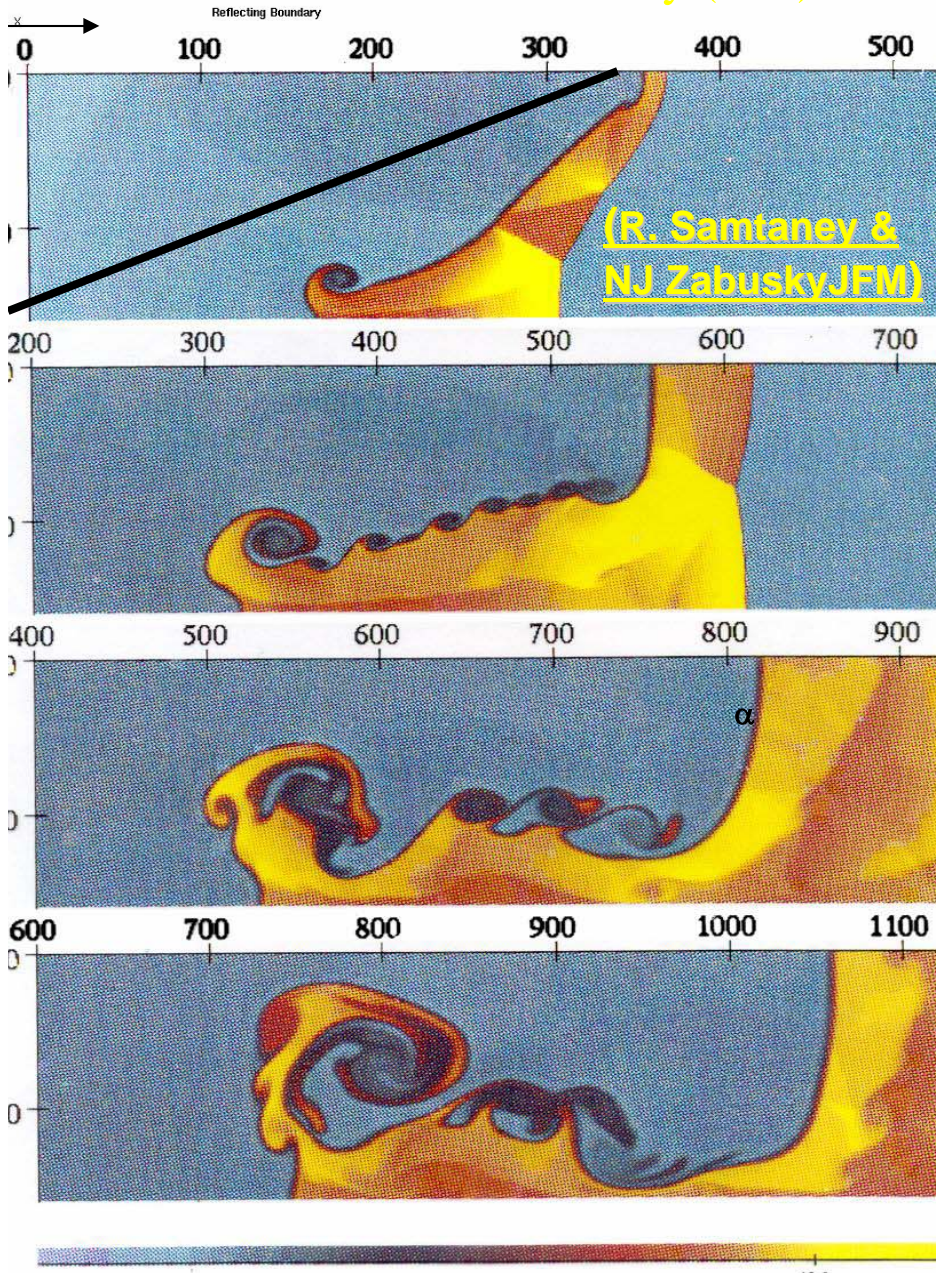
Density: Shock inclined interface (Samtaney '96)



Richtmyer-Meshkov Planar inclined Interface

Density (left)

Vorticity (right)

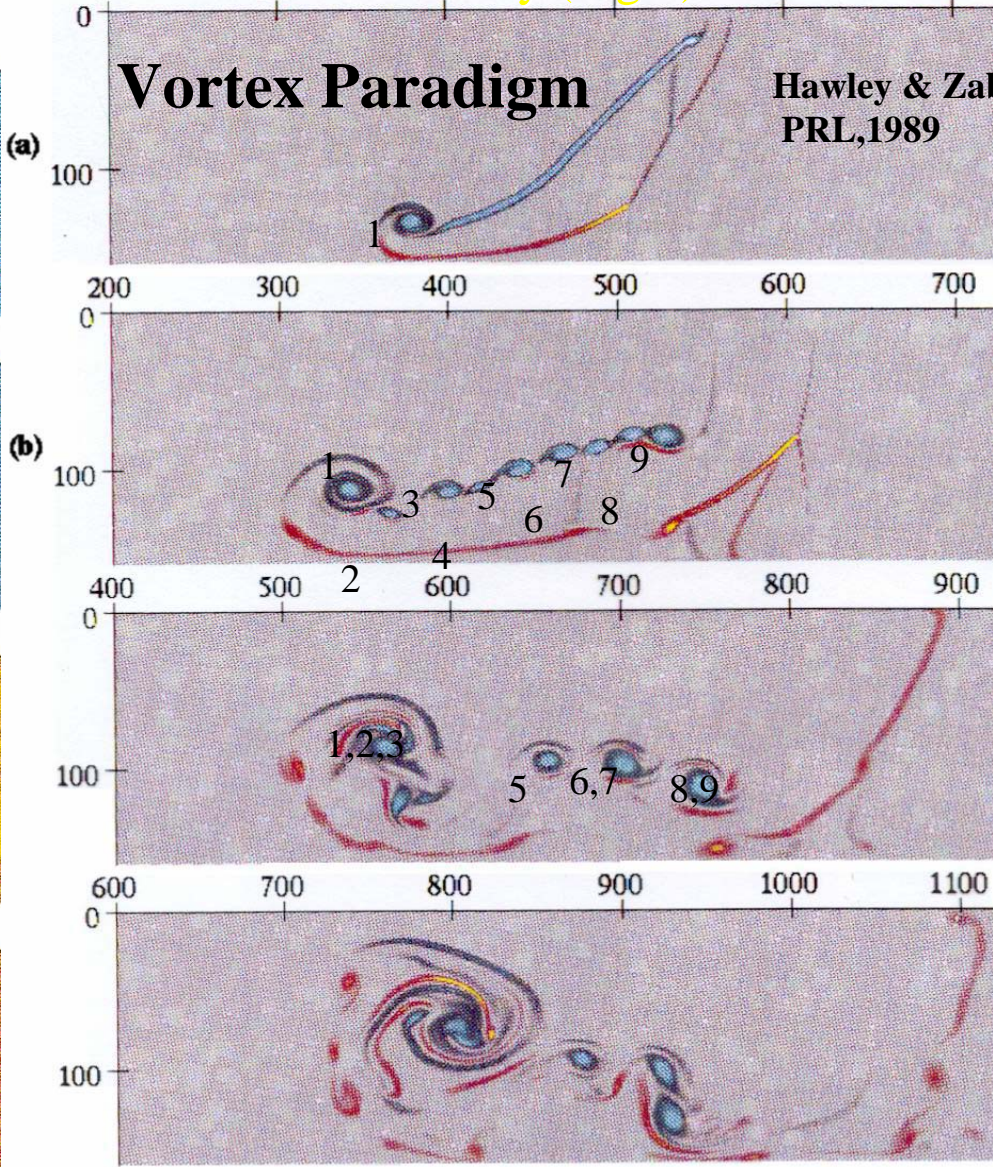


(R. Samtaney & NJ Zabusky JFM)

Density $M=2.0$, $E=3.0$, $A=60$; $a=41.40$ $b=83.00$ $c=139.10$ $d=1$

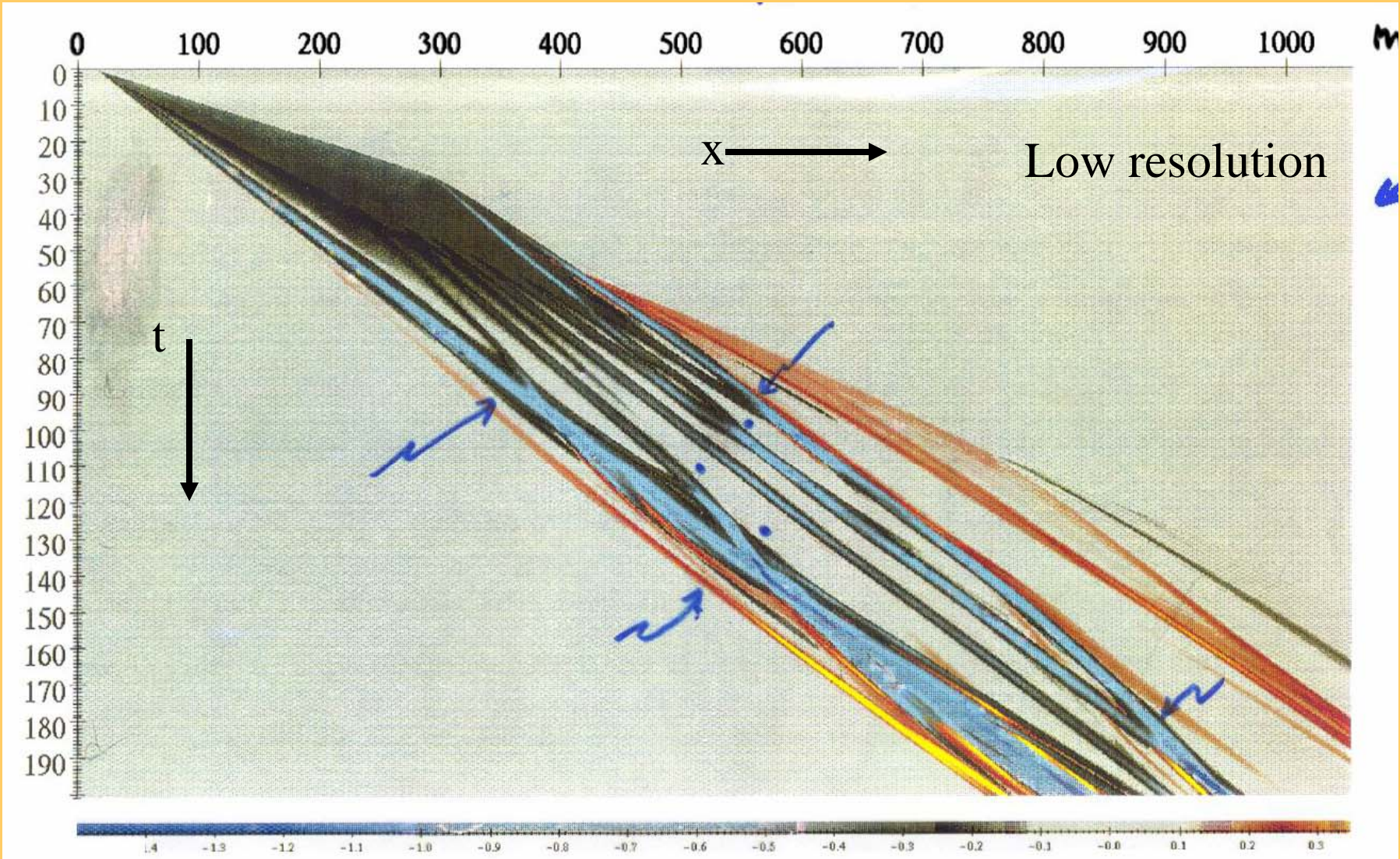
Vortex Paradigm

Hawley & Zabusky PRL, 1989



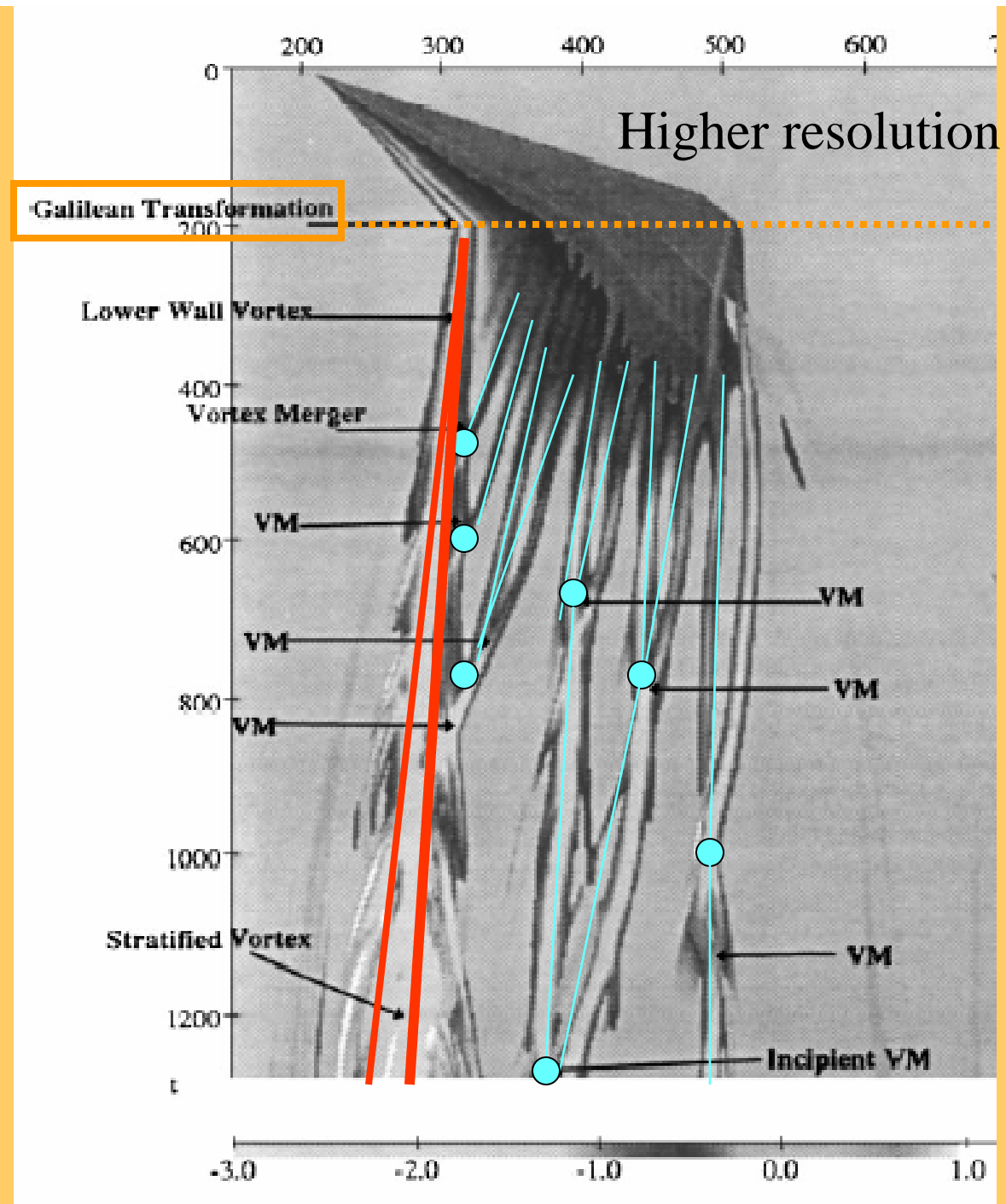
Vorticity $M=2.0$, $E=3.0$, $A=60$; $a=41.40$ $b=83.00$ $c=139.10$ $d=1$

Integrated vorticity
space- time diagram, $M=2$; $A=0.5$

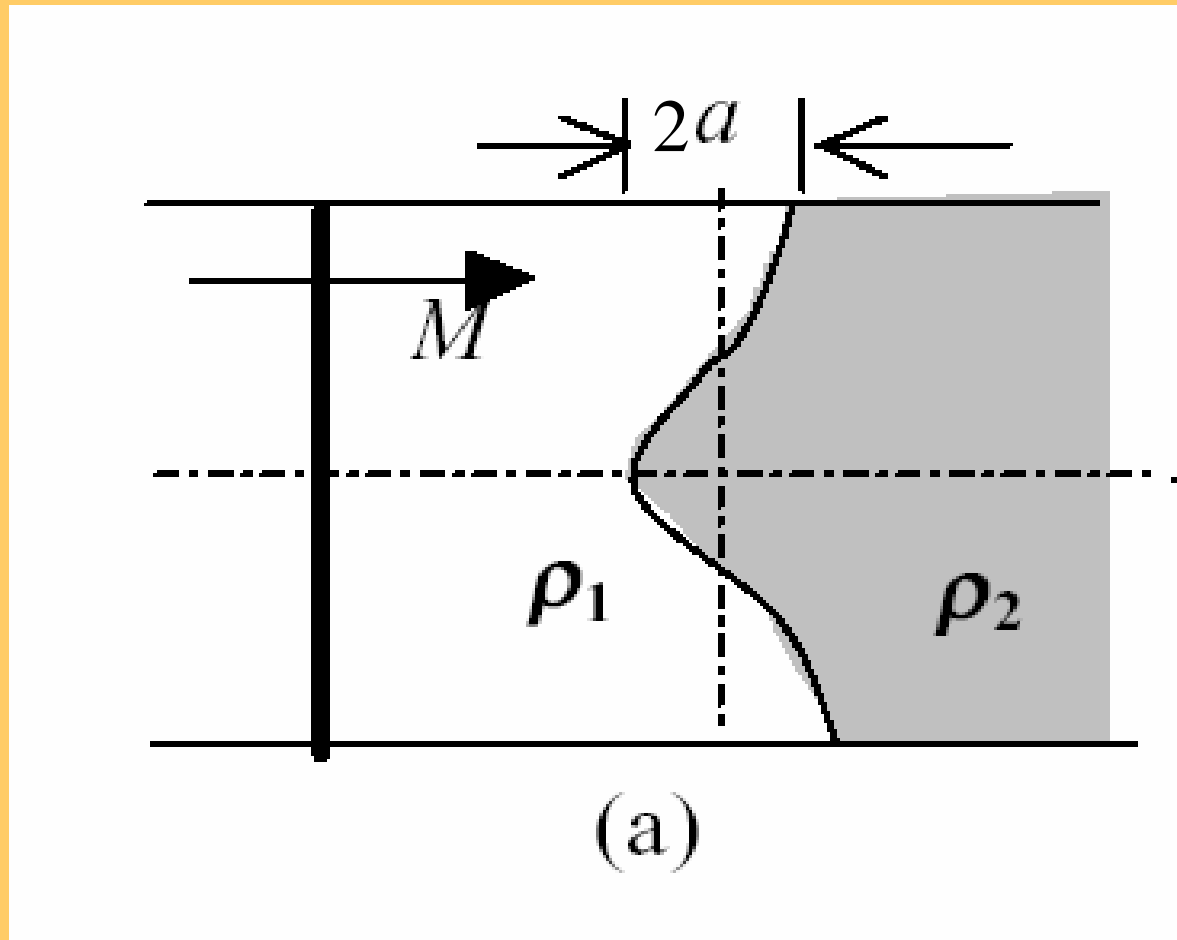


$M=2.0$ $E=3.0$ $SA=60$ Composite Space Time Diagram: $\mu=1.5$

y-
integrated
vorticity
for a shock
interacting with
a planar layer:
($M = 1.5$, $A=0.5$;
[Air/R22], at an
angle of 60
deg)



Classical RM Geometry



$$M = \text{Mach No.}, \quad A = \text{Atwood No.} = \frac{\rho_1 - \rho_2}{\rho_1 + \rho_2}$$

RT & RM Finite-Time singularity

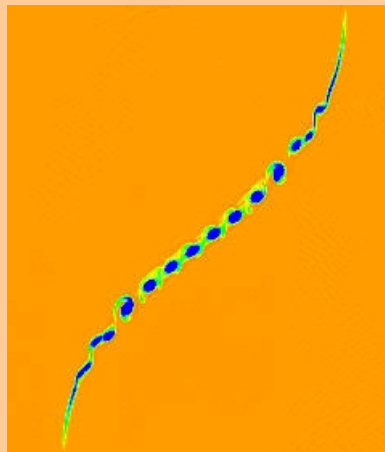
(ill-posed nature of vortex sheet evolution)

Kelvin-Helmholtz of vortex sheets

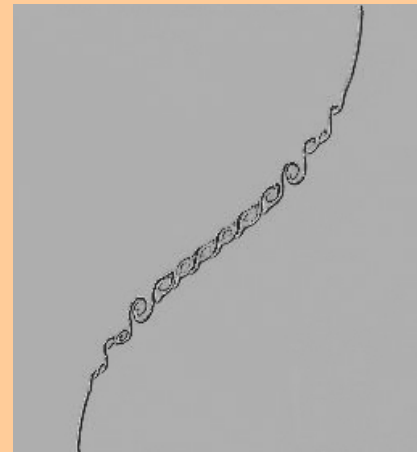
– *Finite-time Moore curvature singularity* for **single sine wave**

– *Regularization* via finite ITL

$$u_t + uu_x = 0$$



Vorticity

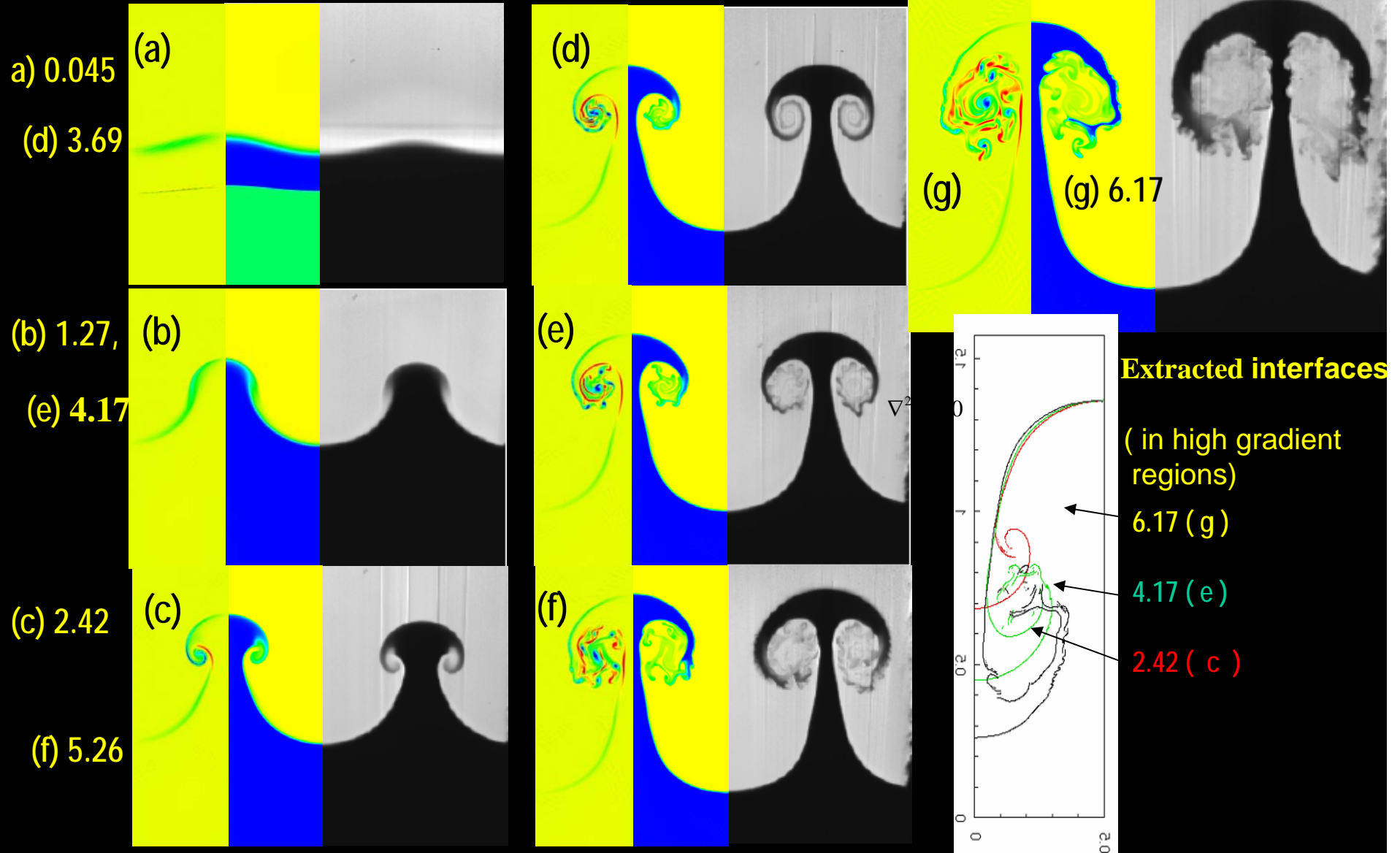


$\nabla^2 \rho$

Sinusoidal interface

Vizlab Simulations (PPM) of G. Peng and S. Zhang & Jacob & Krivets' Experiment (PLIF)

($M = 1.3, A = 0.635$) Juxtaposition: Columns of *Vorticity*, *Density*, *Experimental PLIF*



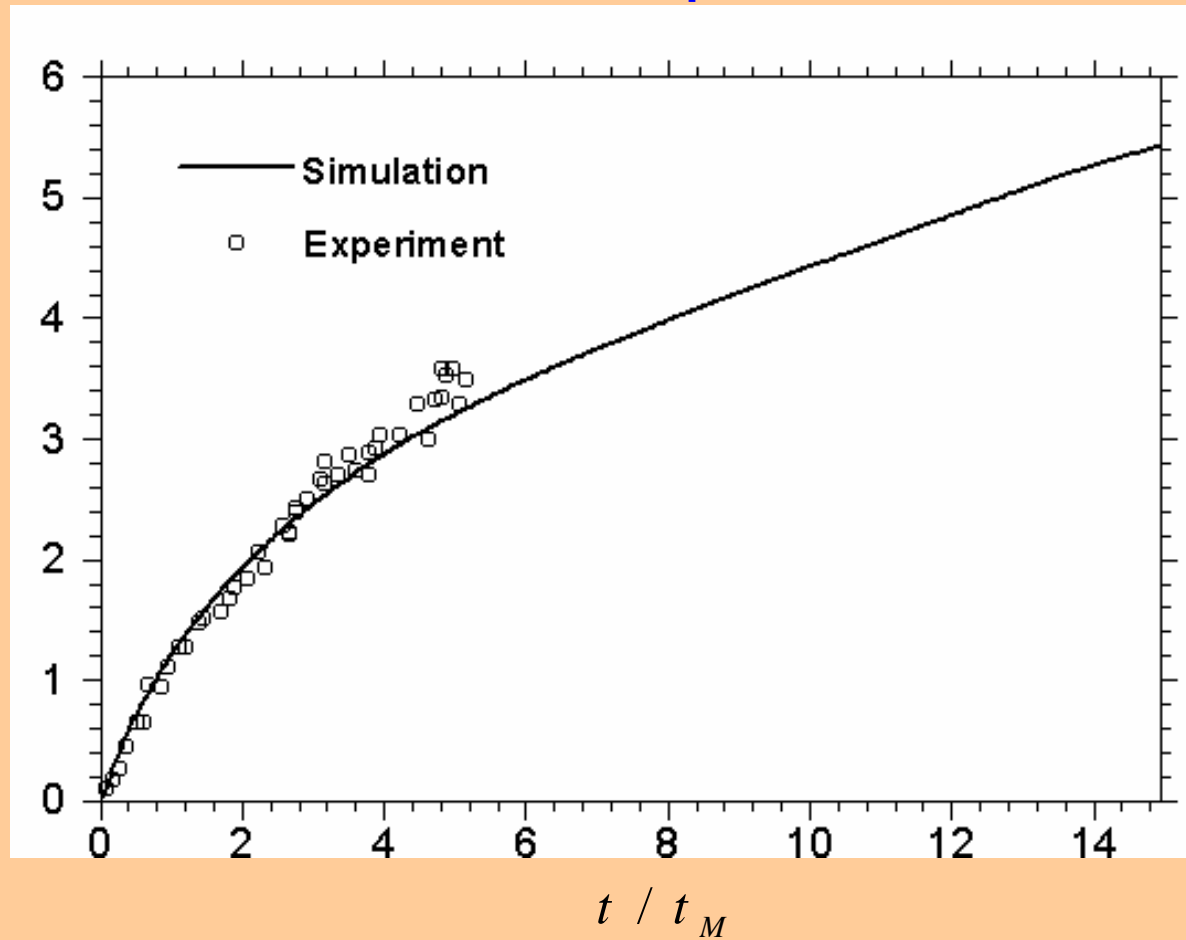
OVERVIEW: “AIFS” Accelerated Inhomogeneous Flows

A Vortex Approach

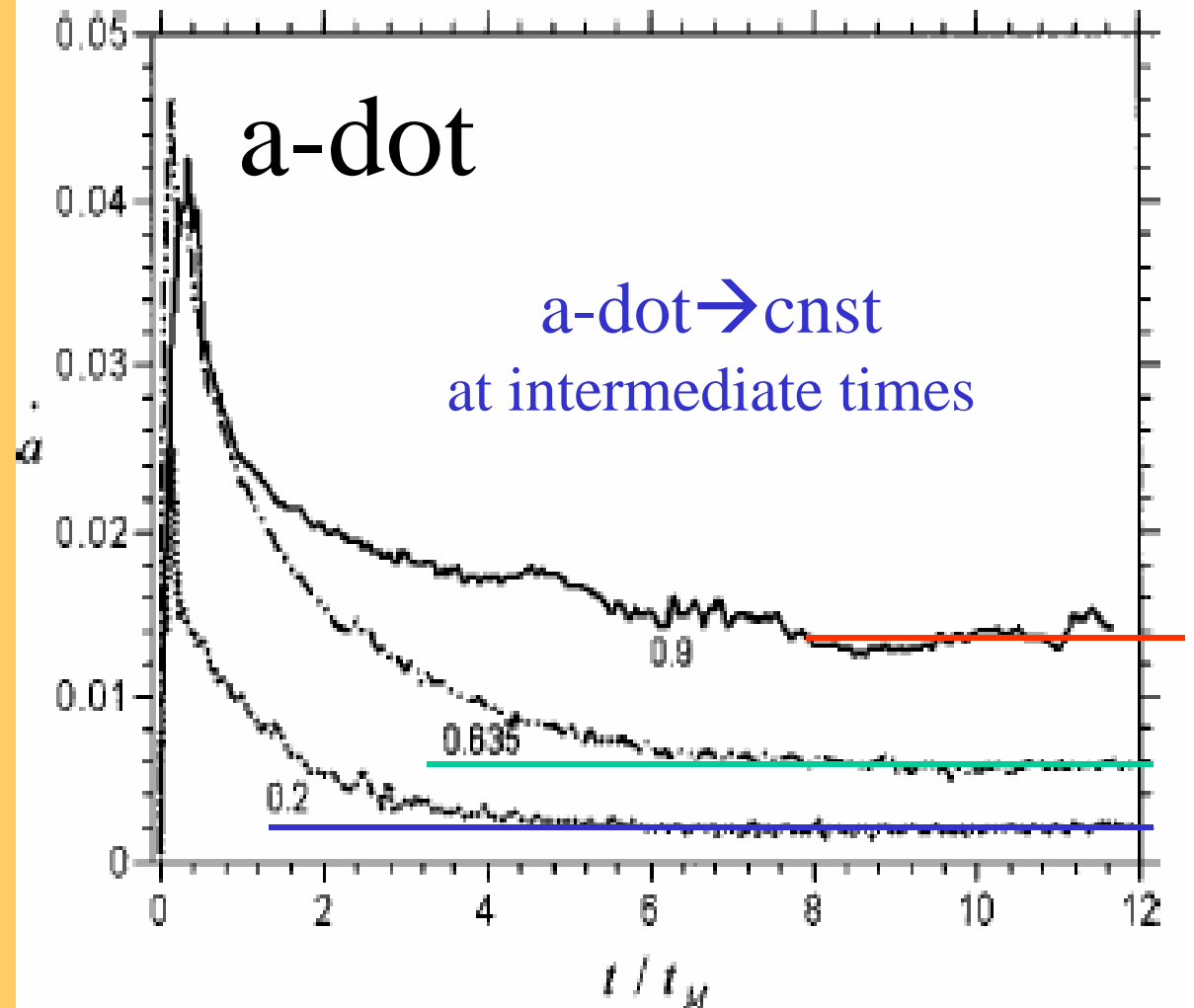
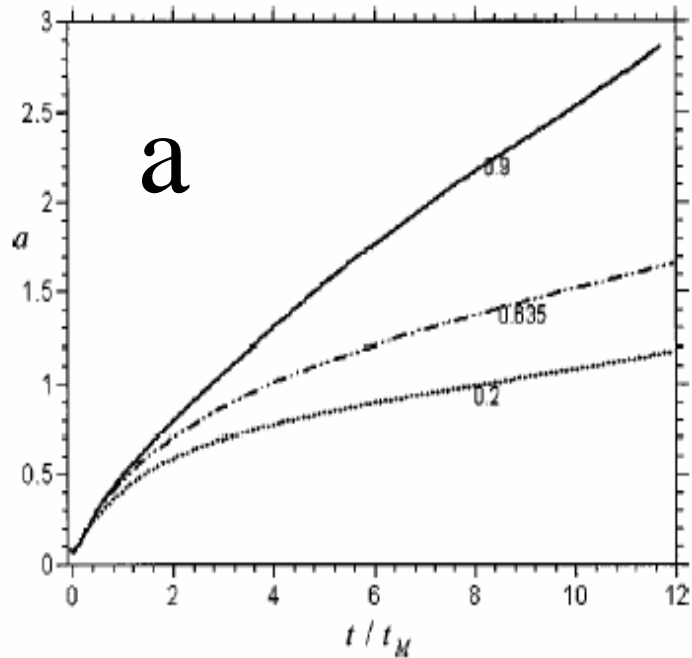
- Topics
 - Well-posedness and finite initial transition layer
 - **RM $a\text{-dot}$ \rightarrow constant at intermediate times**
 -

Amplitude growth $a(t)$: simulation and experiment

$$k(a - a_0^*)$$

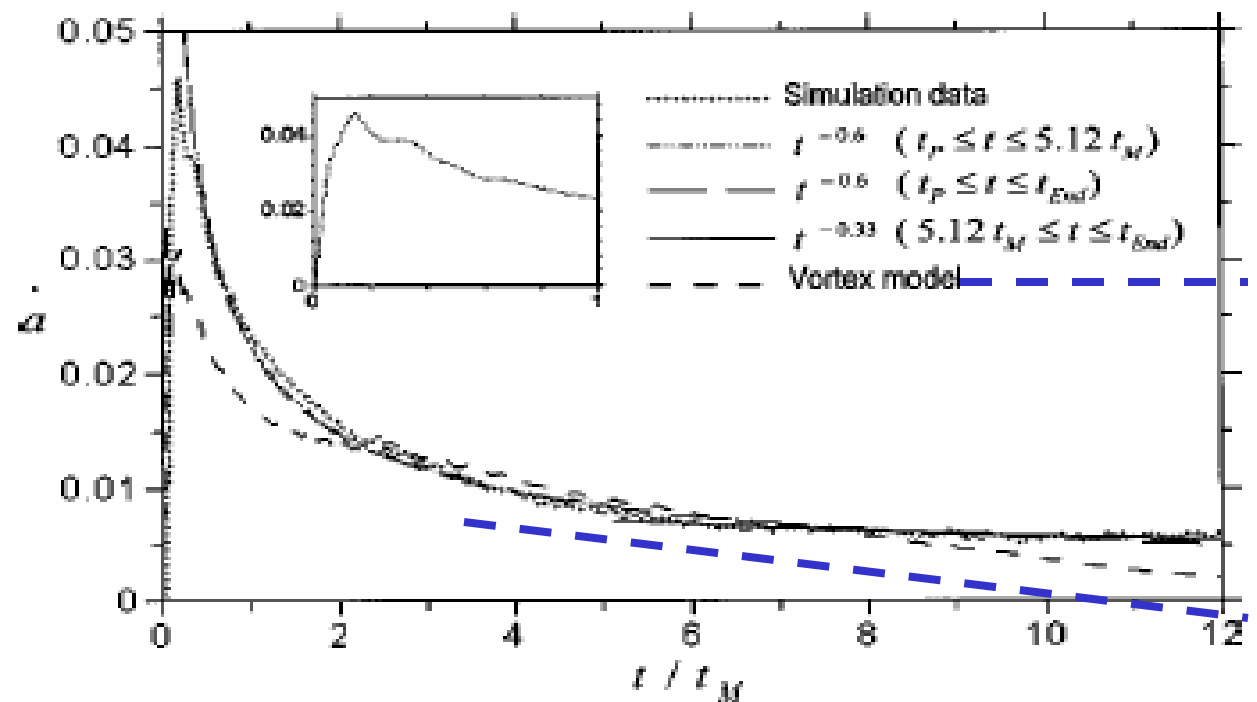


a & \dot{a} for $\Lambda = 0.2, 0.635$ & 0.9



A-dot comparisons: Simulations & proposed *adjusting-one vortex model*

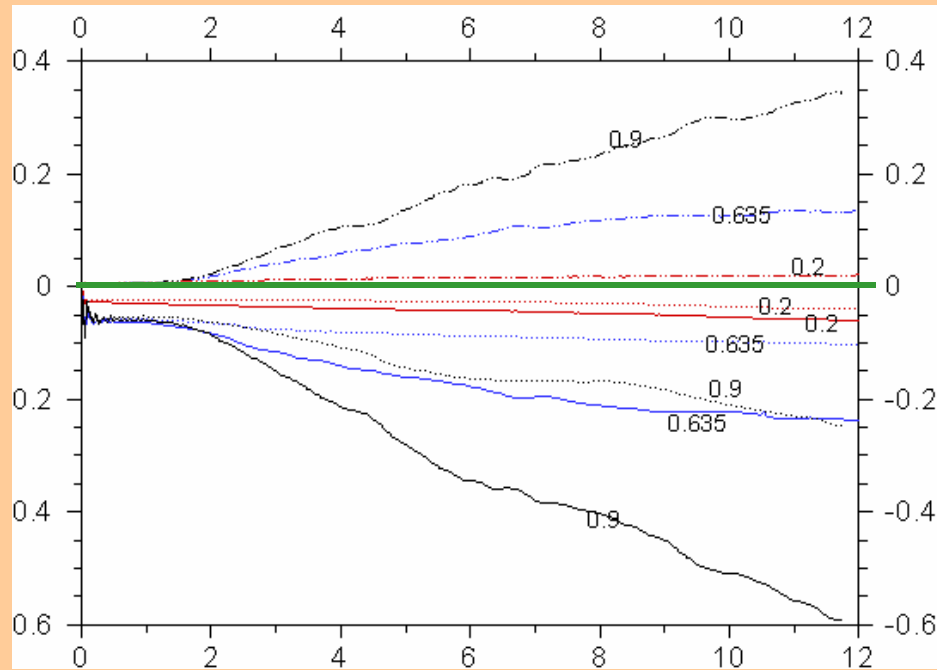
$$\dot{a}_{\text{vortex}} = - \frac{k\Gamma \sin kx_c}{4\pi} \left(\frac{1}{\cosh(kd_s) - \cos kx_c} + \frac{1}{\cosh(kd_b) + \cos kx_c} \right),$$



OVERVIEW:RM

- **Topics**
 - Well-posedness and initial transition layer
 - RM $a\text{-dot}$ -> constant at intermediate times
 - **Circulation generation (*vortex bilayers*) & gradient Intensification**

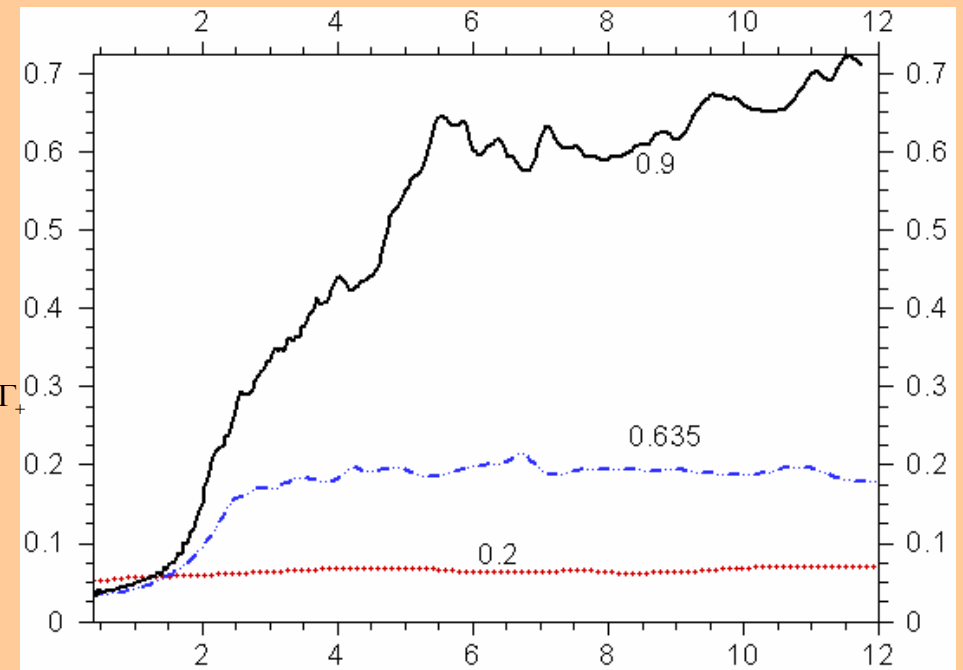
Global quantifications: Circulation & Enstrophy



t / t_M

Circulation $\Gamma_+, \Gamma_-, \Gamma = \Gamma_+ + \Gamma_-$,

$$\int_D [\omega, \omega_{\pm}](x, y, t) dx dy$$

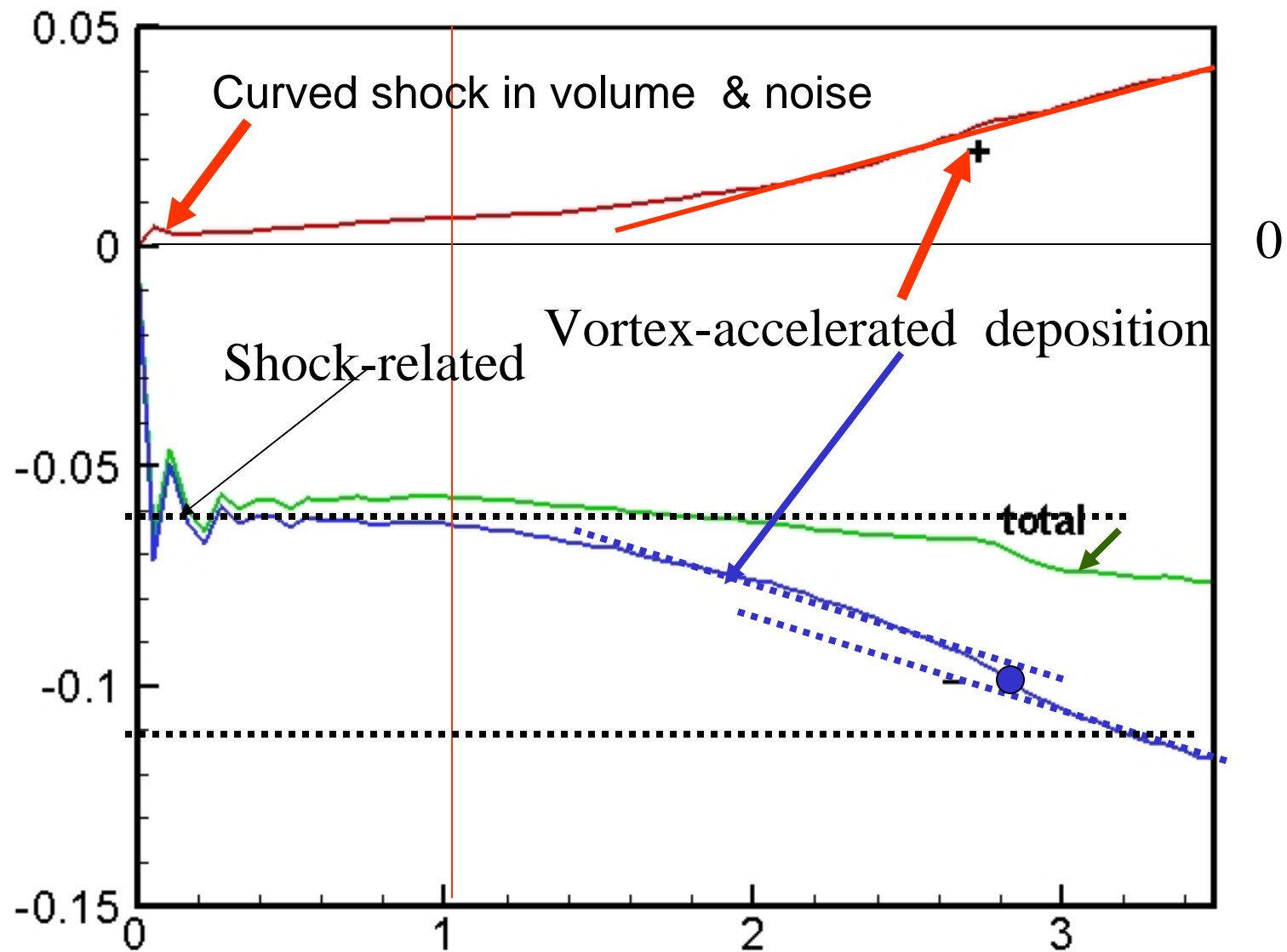


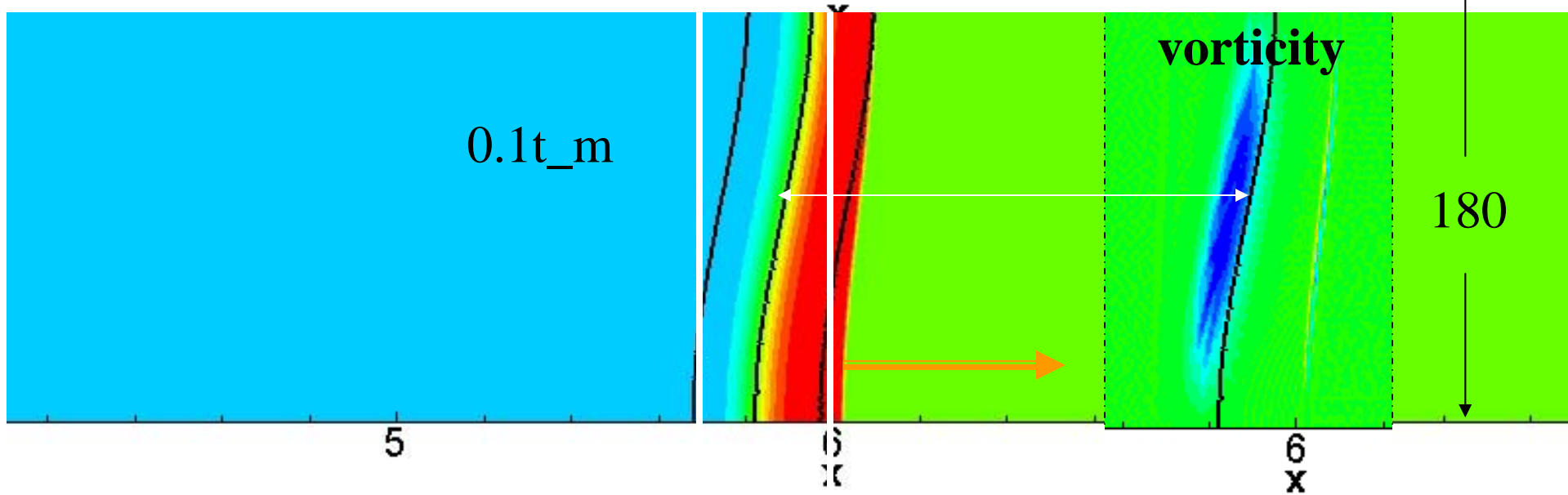
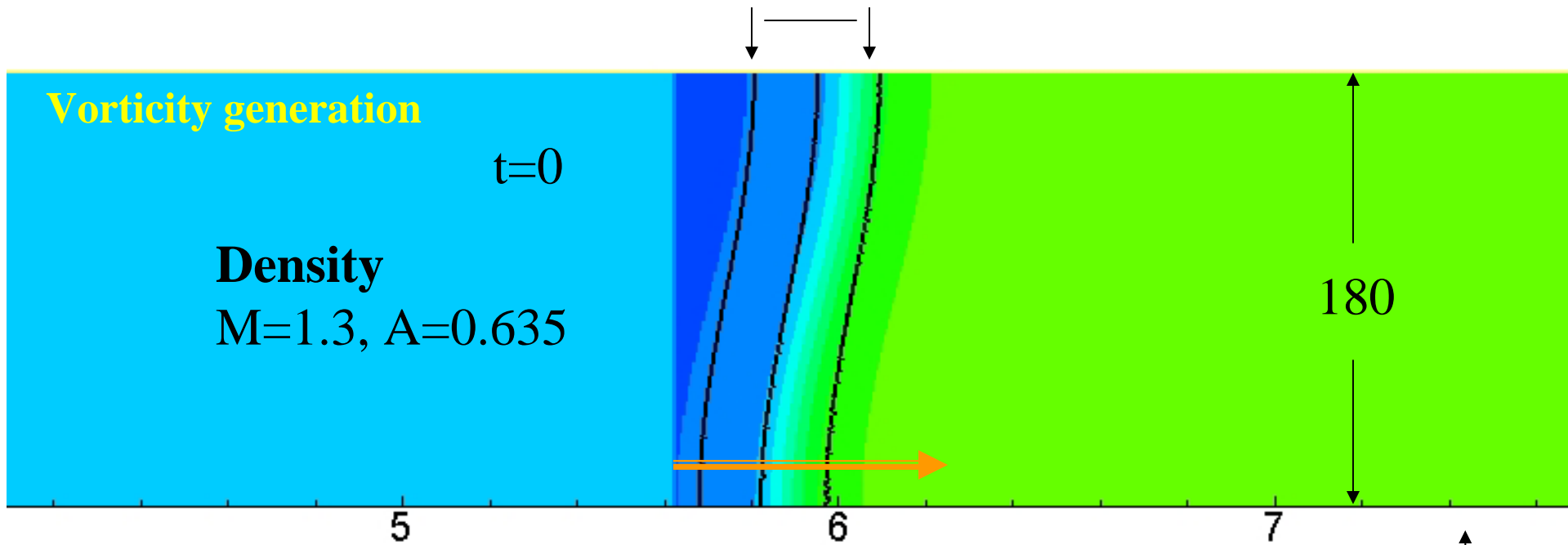
t / t_M

Enstrophy

$$\int_D \omega^2(x, y, t) dx dy$$

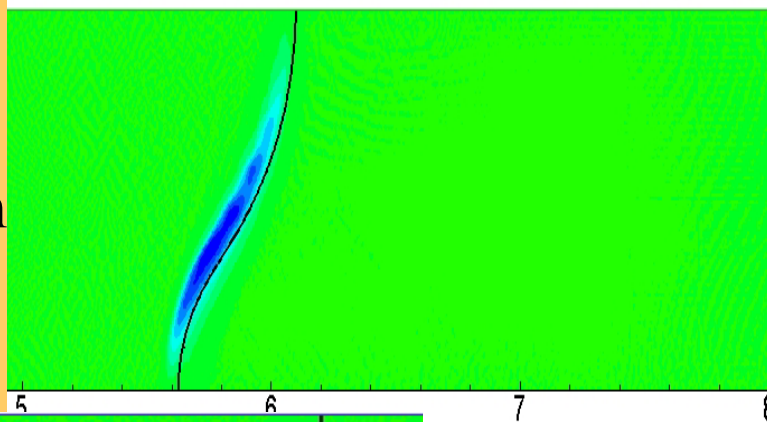
Circulations vs time, RM with $A=0.635$, $M=1.3$



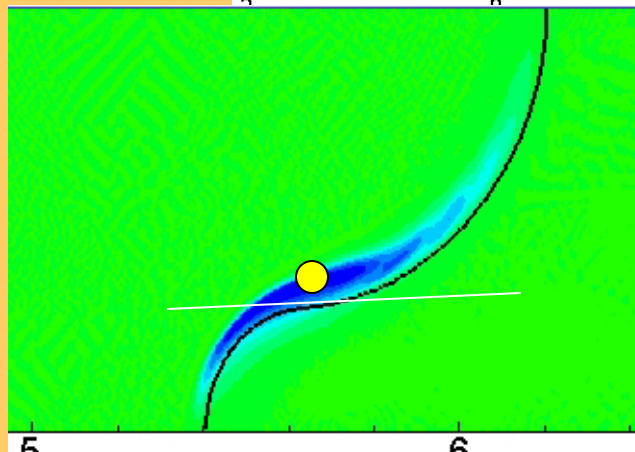


**RM
Baroclinic
vorticity**

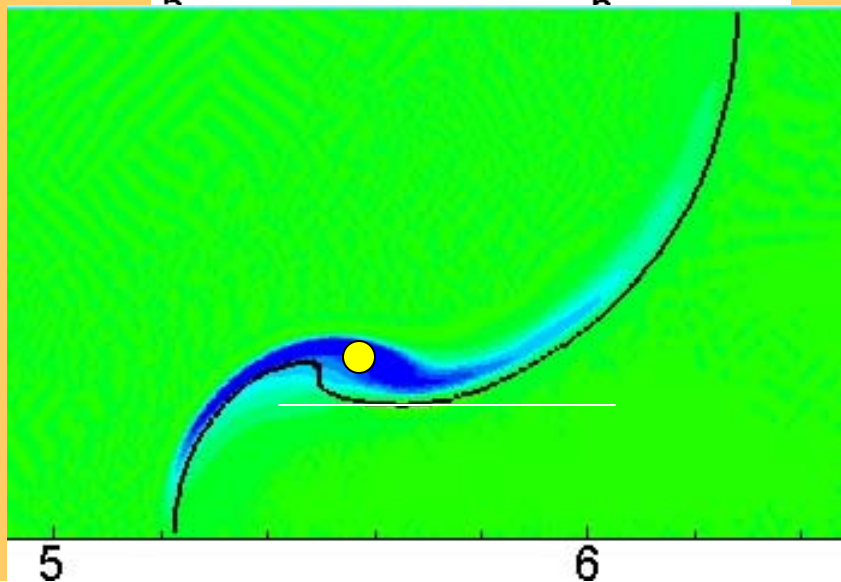
$0.5t_m$



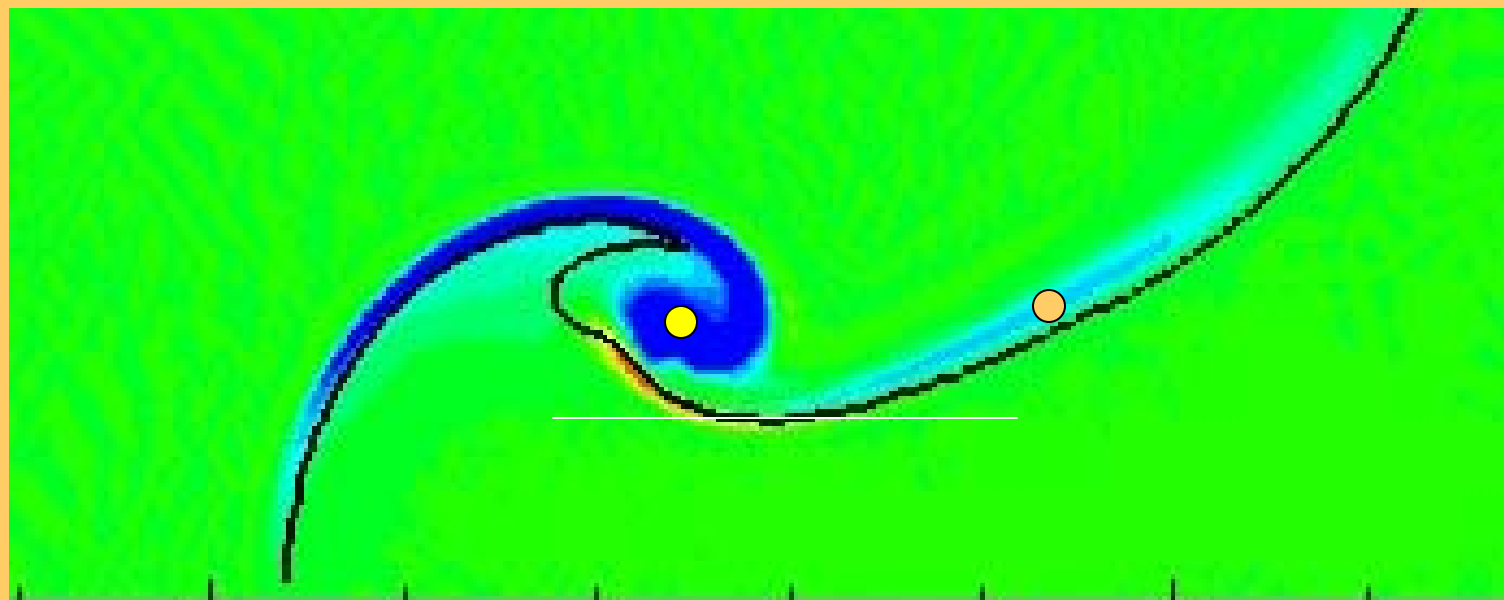
$1.0t_m$



$1.5t_m$

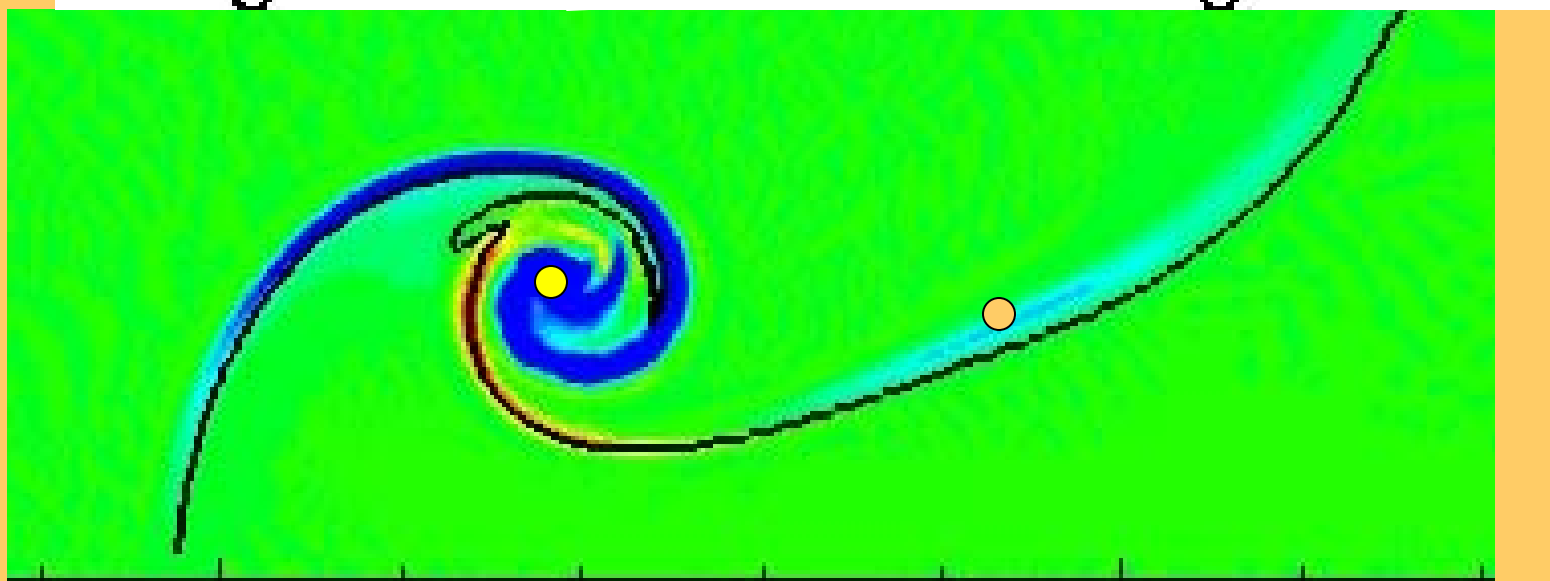


2.0t_m



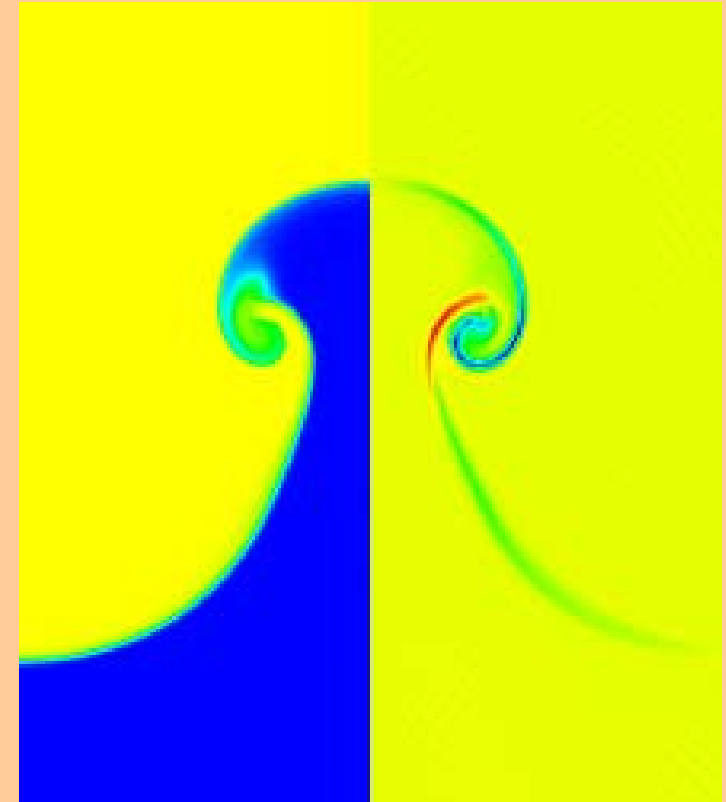
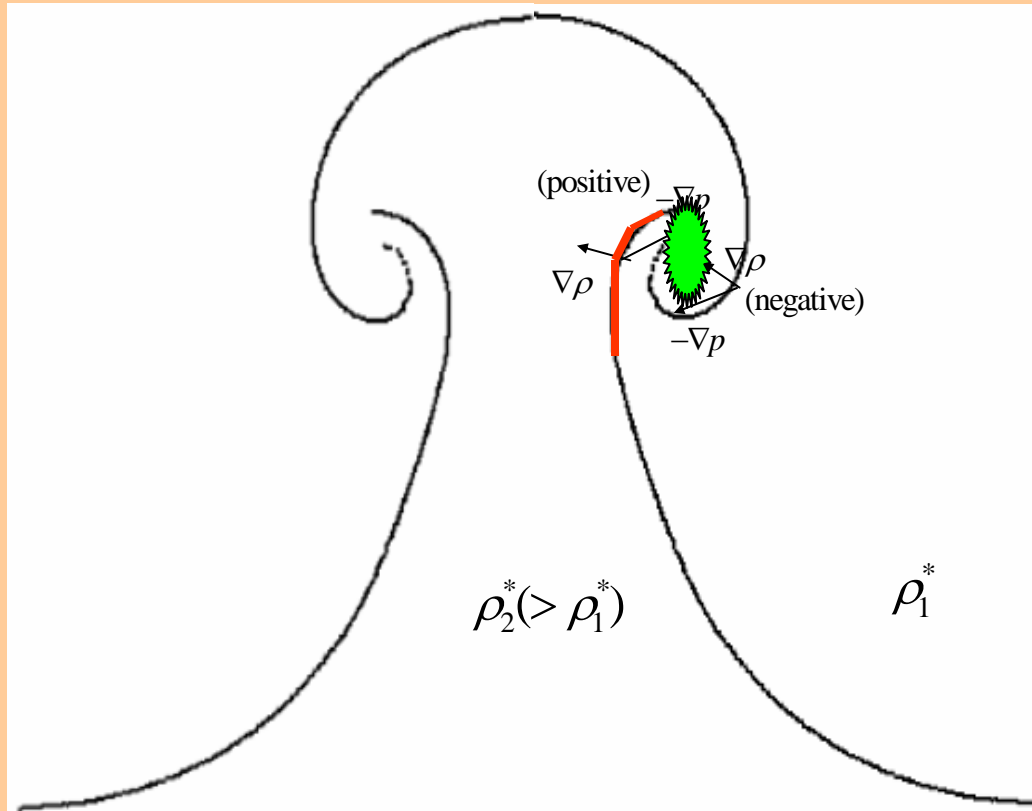
RM
Baroclinic
vorticity

2.5t_m



x

Vortex-accelerated “secondary” baroclinic vorticity deposition



$$\frac{D\omega}{Dt} = \frac{\nabla \rho \times \nabla p}{\rho^2} + \omega \cdot \nabla \mathbf{u} - \omega \nabla \cdot \mathbf{u}$$

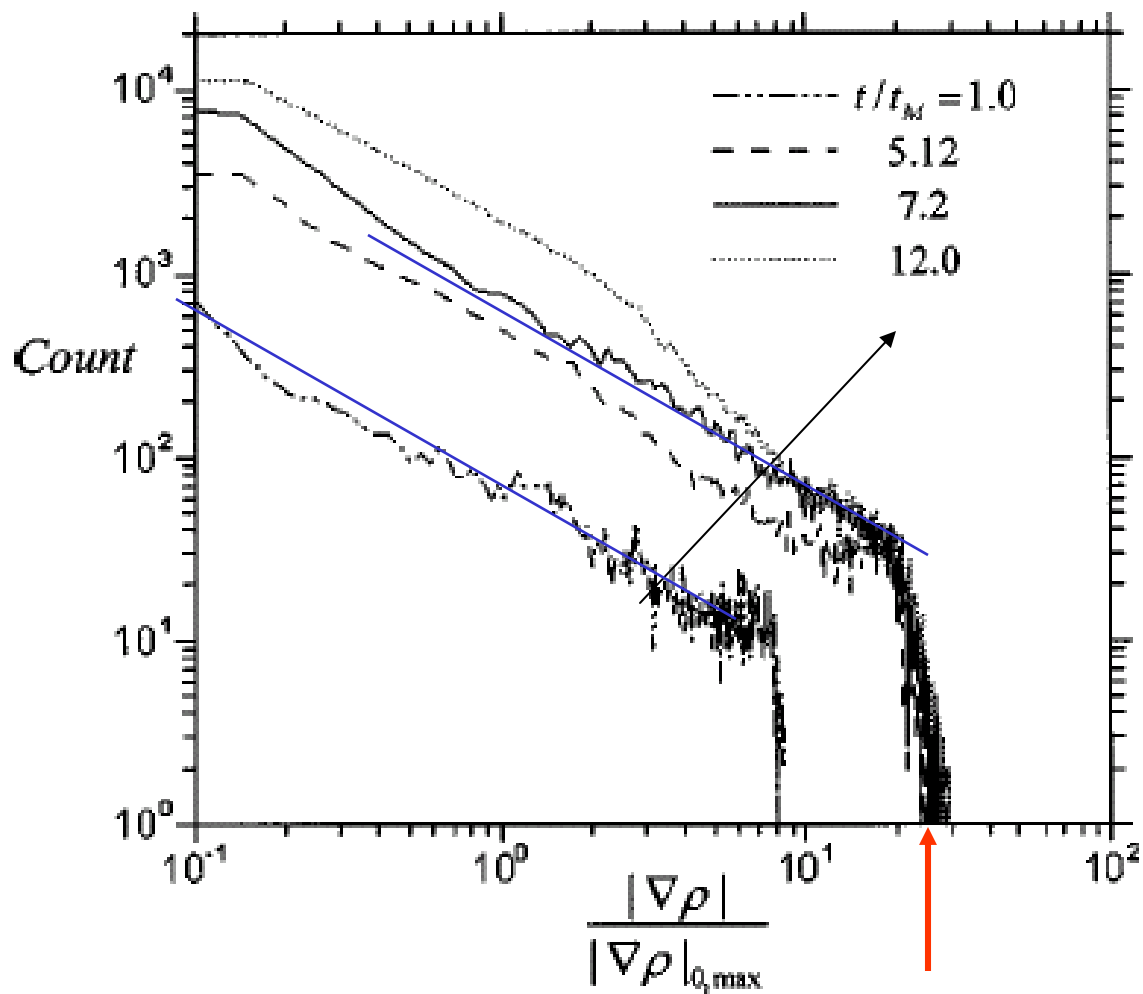
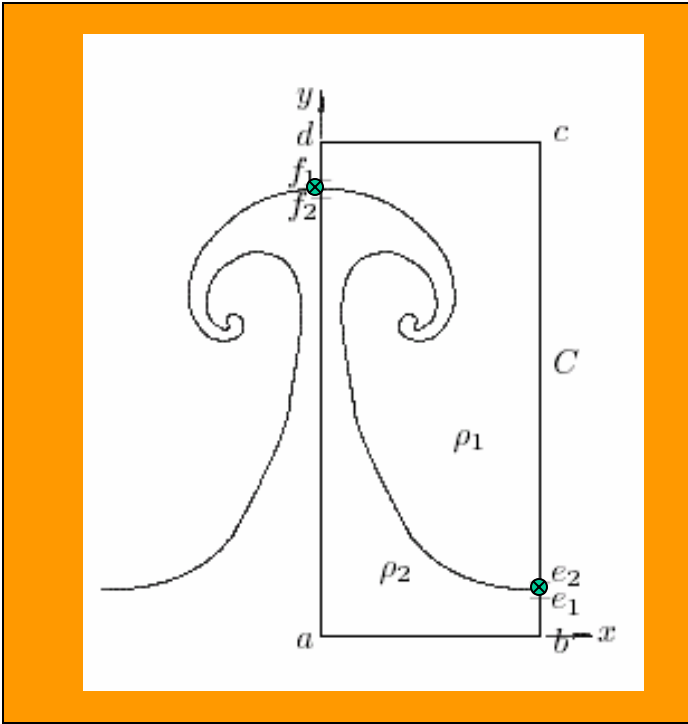


FIG. 13. Density gradient magnitude (normalized by the preshock initial maximum density gradient magnitude) distribution for $A^* = 0.635$ at times $t/t_M = 1.0, 5.12, 7.2,$ and 12 .



Definitions:

$$\mathbf{u} = u\mathbf{e}_x + v\mathbf{e}_y \quad \boldsymbol{\omega} = \omega\mathbf{e}_z = (-\partial_y u + \partial_x v)\mathbf{e}_z,$$

$$\Gamma_D = \iint_D \omega \, dx dy.$$

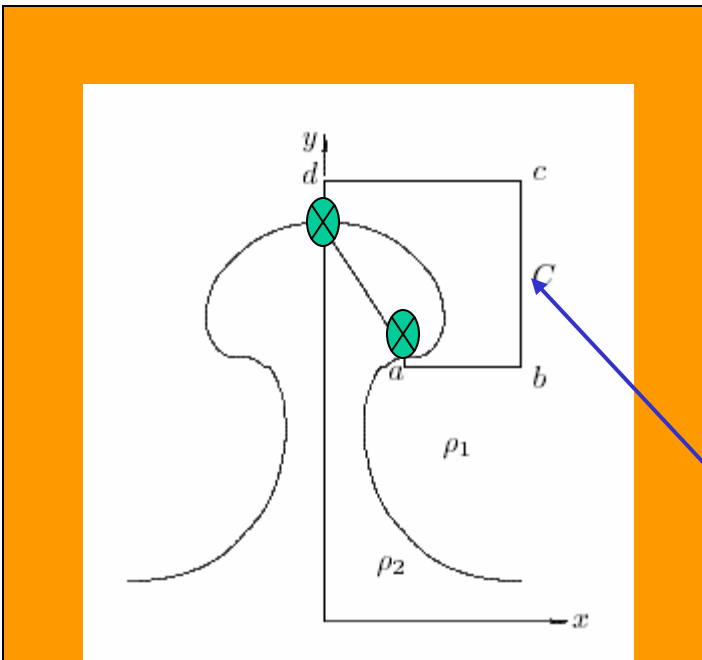
Circulation Rate of Change

$$\partial_t \Gamma = -\frac{1}{2} \oint \nabla(\mathbf{u} \cdot \mathbf{u}) \cdot d\mathbf{s} + \oint \nabla(\mathbf{u} \times \boldsymbol{\omega}) \cdot d\mathbf{s} + \oint \rho^{-1} \nabla p \cdot d\mathbf{s}$$

$$= 0 + 0 + \oint \rho^{-1} dp$$

$$= -\int_{e_1 e_2} p \rho^{-2} d\rho - \int_{f_1 f_2} p \rho^{-2} d\rho$$

$$\partial_t \Gamma = -(p_b - p_t) \left(\frac{1}{\rho_1} - \frac{1}{\rho_2} \right) = -(p_b - p_t) \frac{(\rho_2 - \rho_1)}{(\rho_2 \rho_1)}$$



Choose domain to select positive or negative, etc

Circulation Data : Computing & Filtering

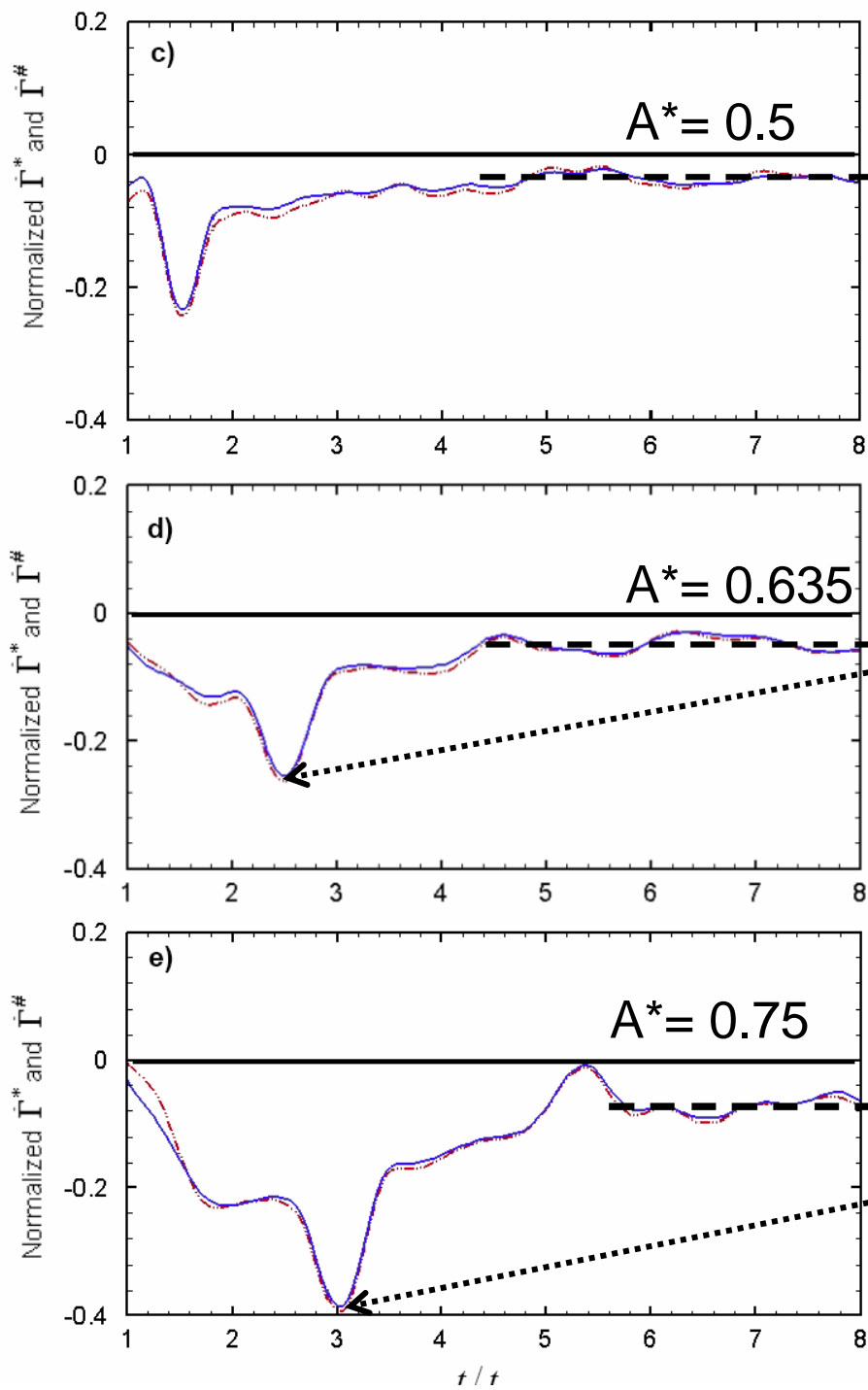
$$\dot{\Gamma} \approx - \int_{a1}^{a2} p(d\rho / \rho^2) - \int_{b1}^{b2} p(d\rho / \rho^2), \quad (3)$$

$$\tilde{f}_{i,j}(t_n) = \sum_{t_m = -t_M/6}^{t_m = +t_M/6} f_{i,j}(t_m) \Phi(t_n; t_m)$$

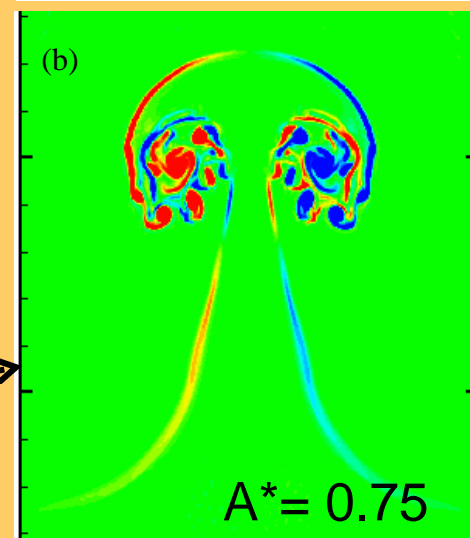
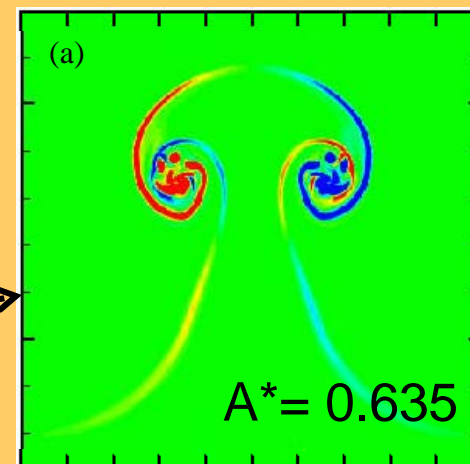
where $\Phi(t_n; t_m) = T^{-1} [1 + \cos 2\pi(t_n - t_m)/T]$, $-(T/2) \leq t_m \leq (T/2)$, & $T = t_M/3$.

$$\dot{\Gamma}^*(t_n) \equiv \frac{\tilde{p}_a(t_n) - \tilde{p}_b(t_n)}{(\rho_2 - \rho_1) / \rho_2 \rho_1} \quad (4)$$

$$\dot{\Gamma}^\#(t_n) \equiv \sum_D^2 h \left[\frac{(\tilde{\omega}_{i,j}(t_{n+1}) - \tilde{\omega}_{i,j}(t_{n-1}))}{2\delta t} \right]. \quad (5)$$



ω_+ ——— (red line)
 ω_- ——— (blue line)



Summary: 2D Vortex paradigm for the evolution of RM

interfaces through intermediate times

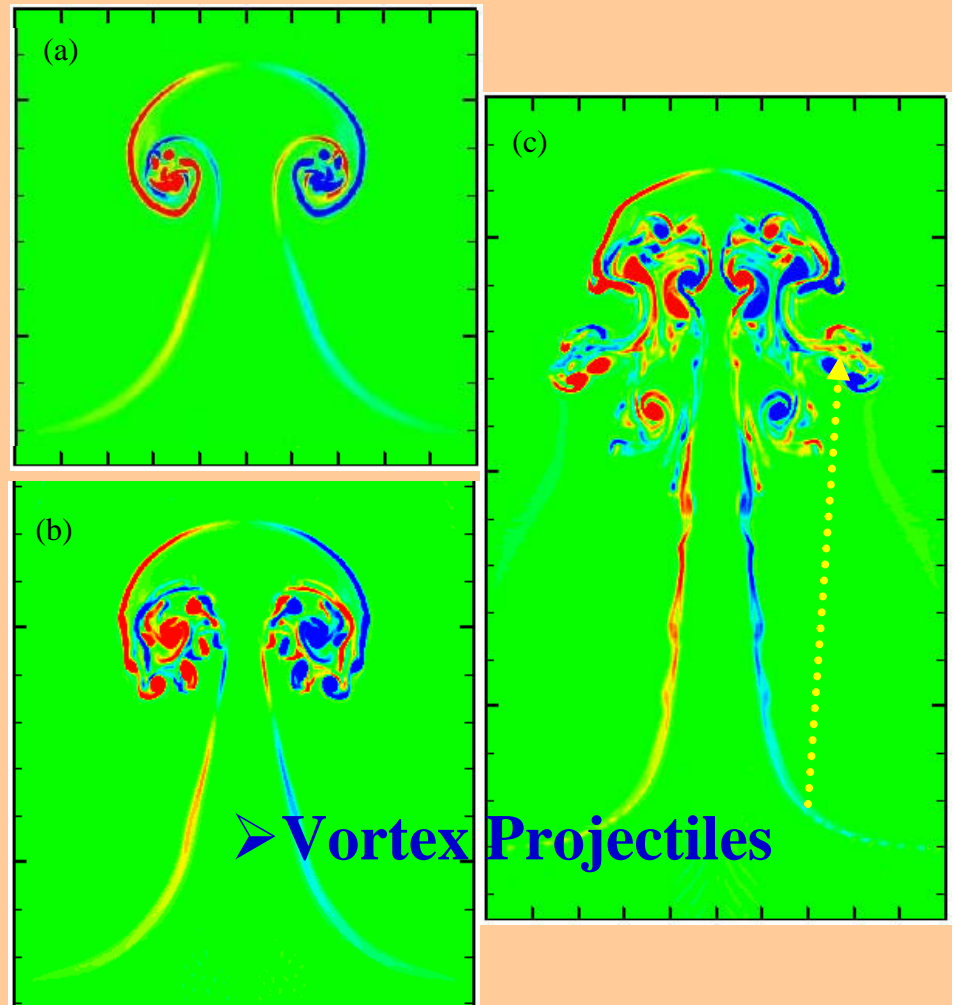
- Intermediate-time dominance of vortex-accelerated vorticity deposition (VAVD) process; [Peng *et al* (2003)]
- Quantification procedures & formulas for net circulation rate of change, $\dot{\Gamma}_D$, comprise a vortex paradigm for the evolution of RM & RT interfaces through intermediate times. [Lee, Peng & Zabusky (Sept. 2006)].
- Special features observed [Lee, Peng & Zabusky (2006)] are signatures of physically important phenomena and include:
 - Gradient intensification of interfacial transition layer
 - Ndp for $t/t_M > 1$ that increases with A^* and the
 - $\dot{\Gamma}_D$ scaling at intermediate times: near-constant negative values that increase with A^* , for $0.5 < A^* < 0.75$.
- Generalization to other accelerated inhomogeneous flow configurations (axisymmetry, shock-cylinder [S. Zhang, N. J. Zabusky, G. Peng, and S. Gupta, PoF, 2004], etc)

OVERVIEW: "AIFS" - RM

➤ Topics

- Well-posedness and initial transition layer
- RM $a \cdot \dot{a}$ -> constant at intermediate times
- Circulation generation (*vortex bilayers*)
gradient Intensification

➤ Vortex Projectiles



RM New Results

- **Secondary Baroclinic Circulation is much greater than Primary (Deposited by Shock).** *PoF '03: G. Peng , S. Zhang & N. Zabusky,*
 - Due to **vortex acceleration & gradient intensification of** transition layer (TL)
- **Vortex Projectiles: Dipolar/Ring-like objects active at all times in determining turbulence and mixing**
- **New diagnostic: Rate of change of circulation in bubble to spike domain.** *PoF, '06 : D.K. Lee, G. Peng, & NJZ*

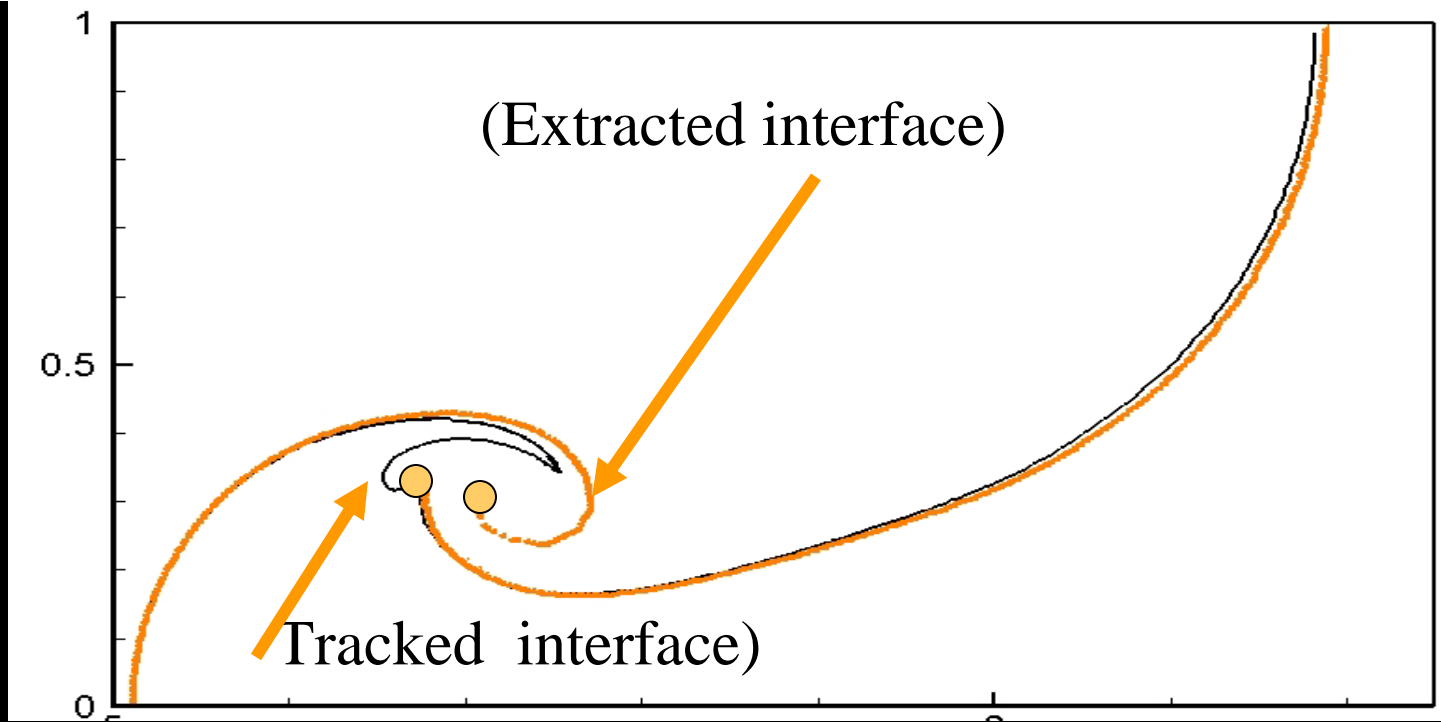
Revisit

Revisit



Revisit

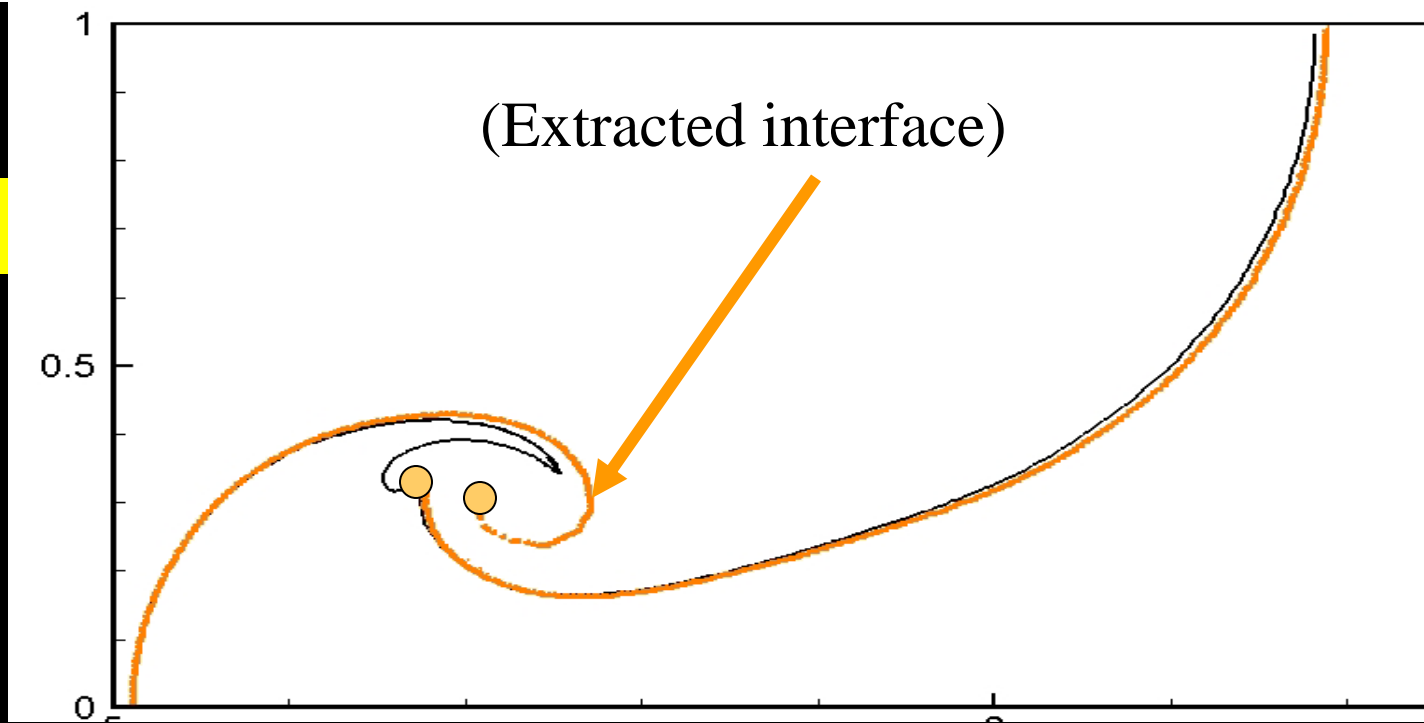
Revisit



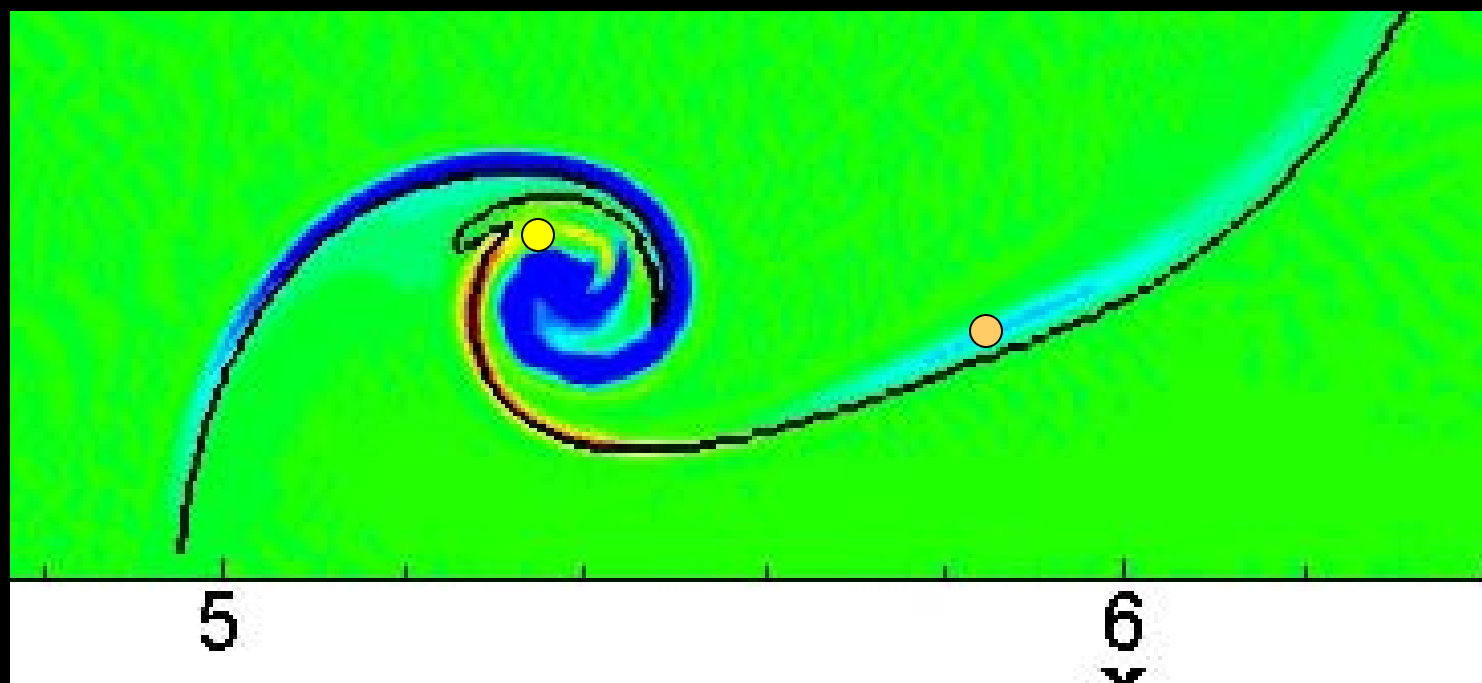
Extraction Algorithm

- $|\text{GRAD rho}| > 0.1 \{ \max |\text{GRAD rho}|_0 \}$
- $\text{LAPLACE}(\text{density}) = 0$

2.2 t_m



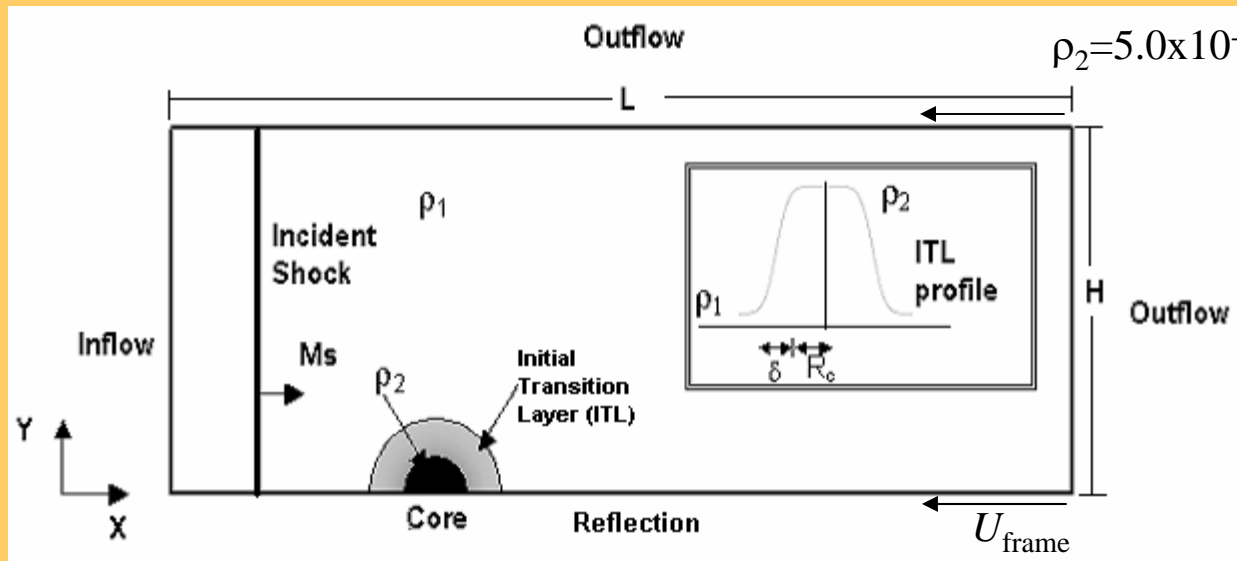
2.5 t_m



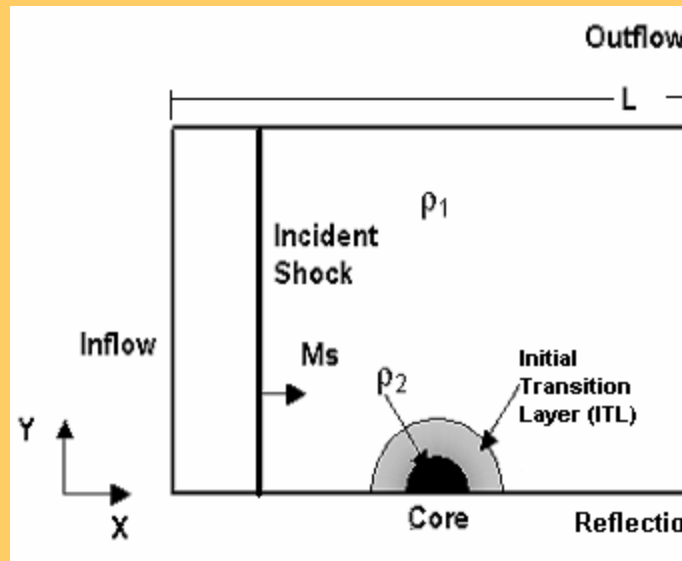
Initial domain

$\rho_1 = 1.0 \times 10^{-3}$ (Air)

$\rho_2 = 5.0 \times 10^{-3}$ (SF_6)



Initial domain



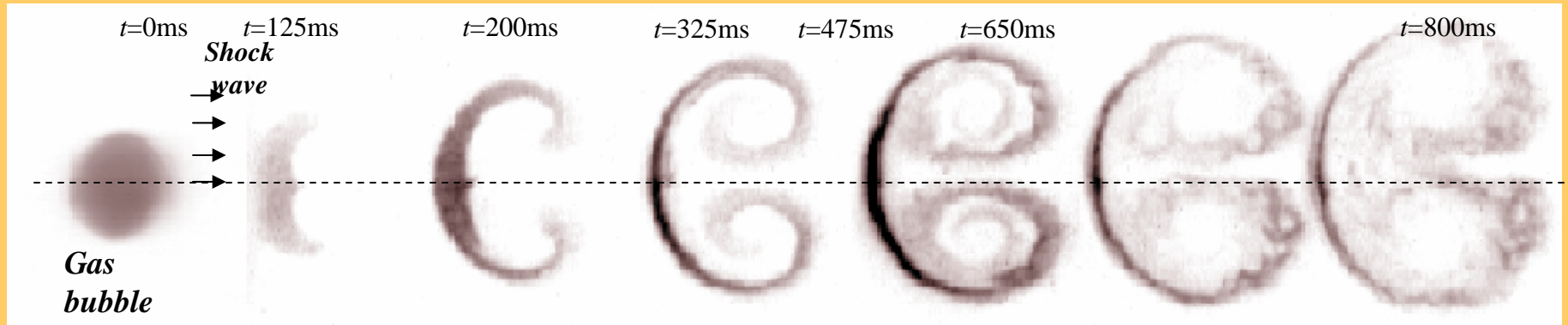
$$\rho_1 = 1.0 \times 10^{-3} \text{ (Air)}$$

$$\rho_2 = 5.0 \times 10^{-3} \text{ (SF}_6\text{)}$$

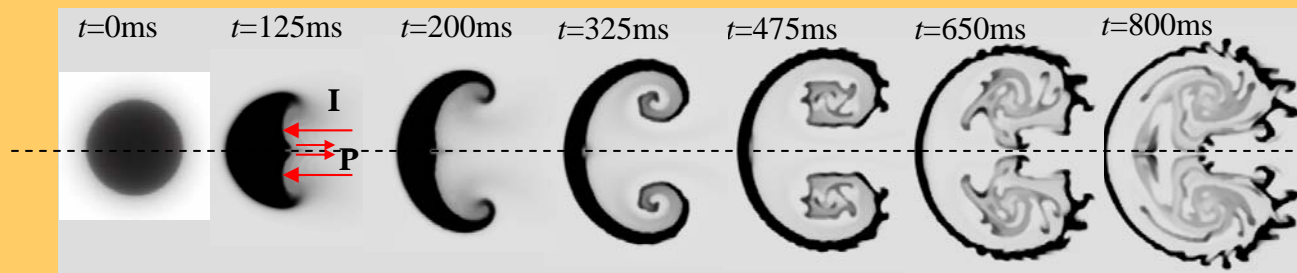


$$U_{\text{frame}}$$

Visiometrics: uncertainty quantification & numerical validation

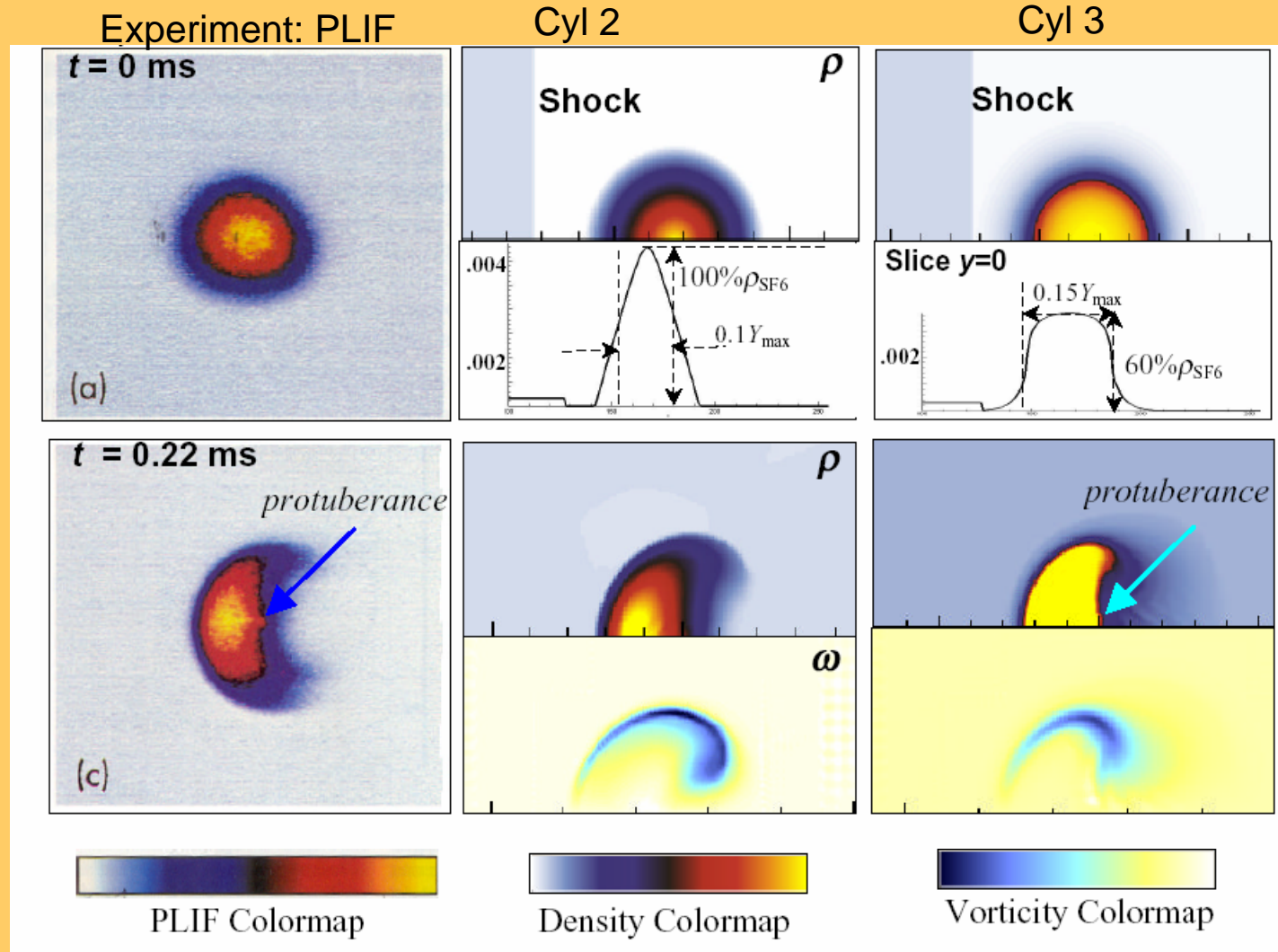


(a). Experimental images (LANL)

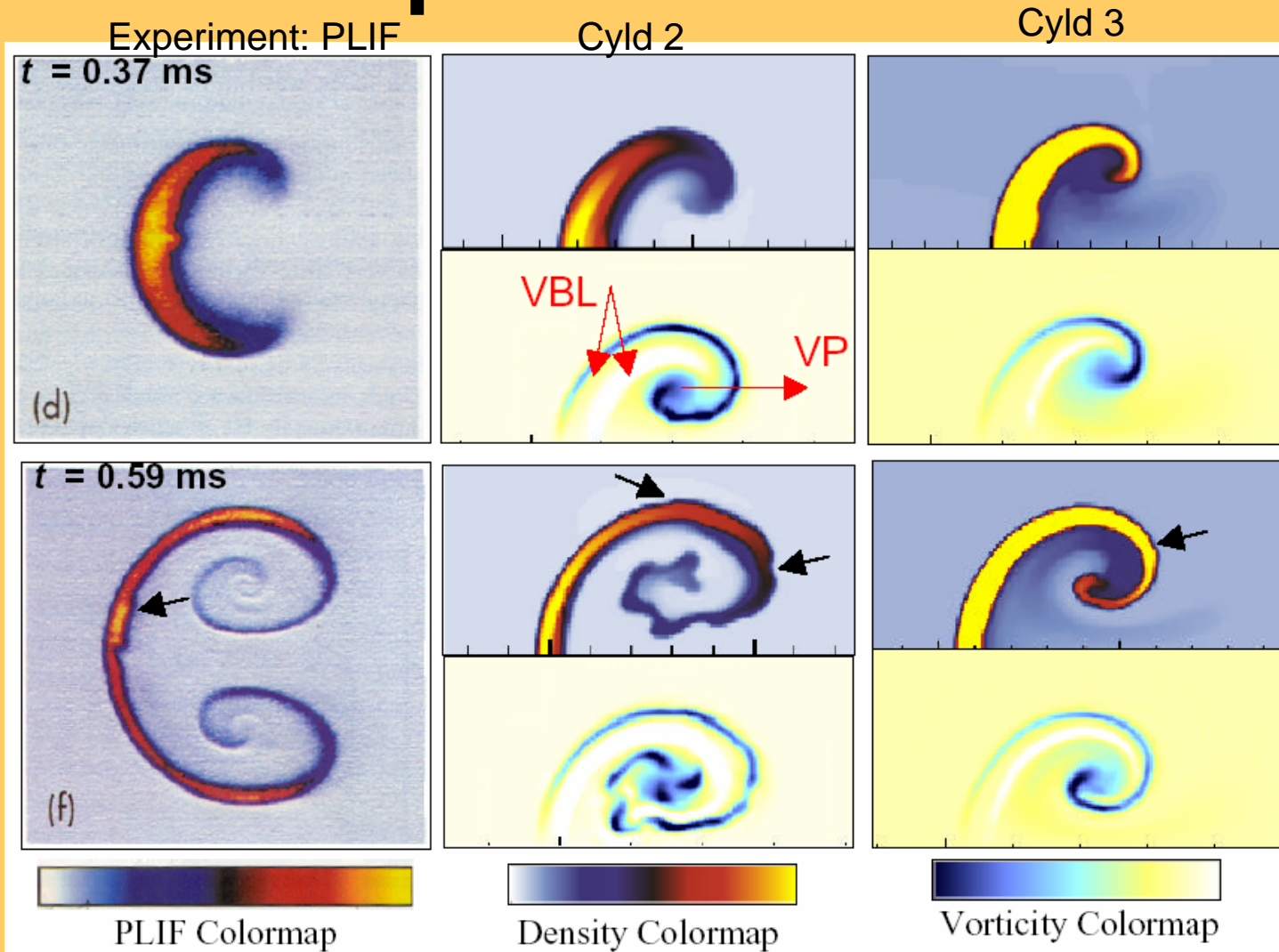


(c). Simulation with visiometrics (SZhang@Vizlab)

Compare Jacobs' Experiment

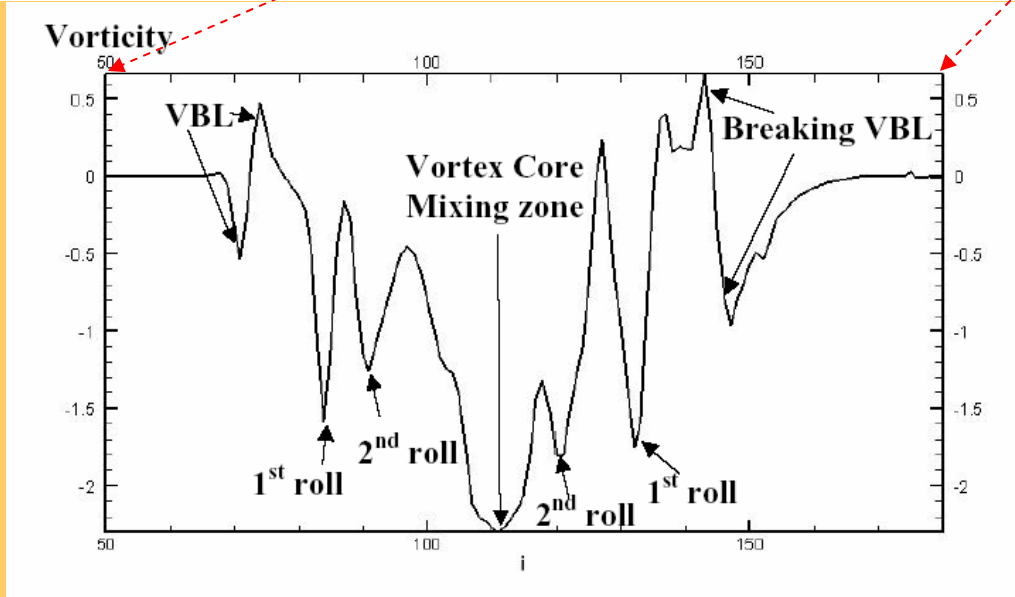
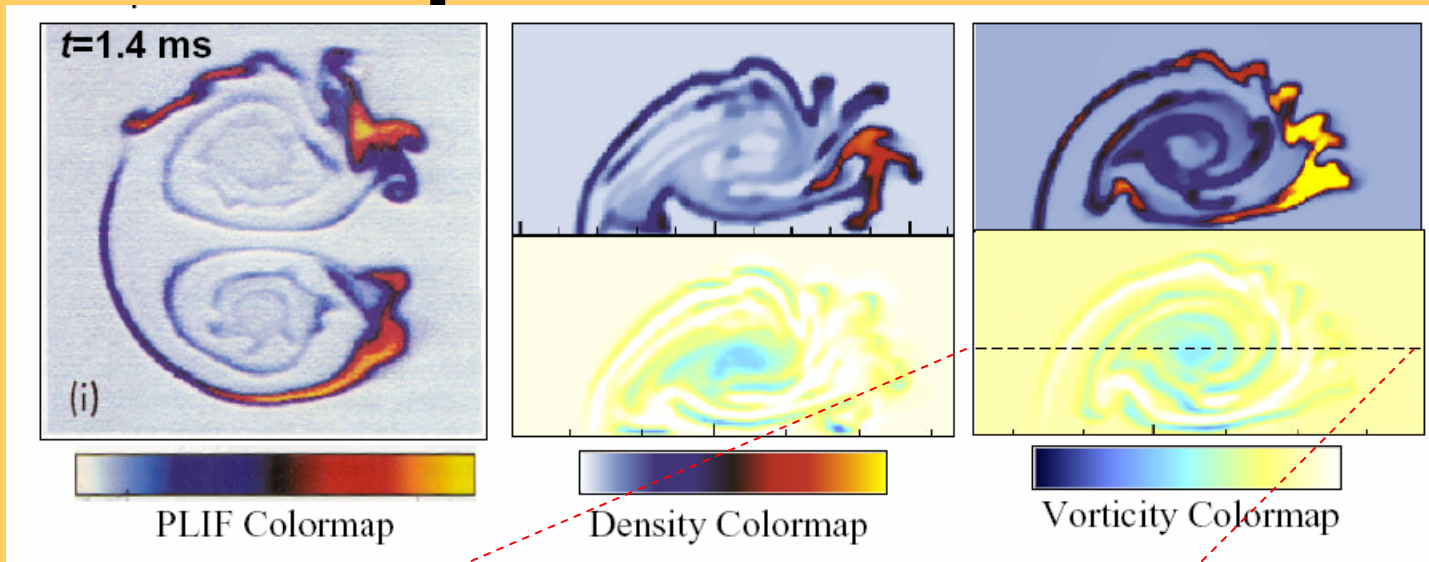


Comparing with Jacobs' Experiment cont.

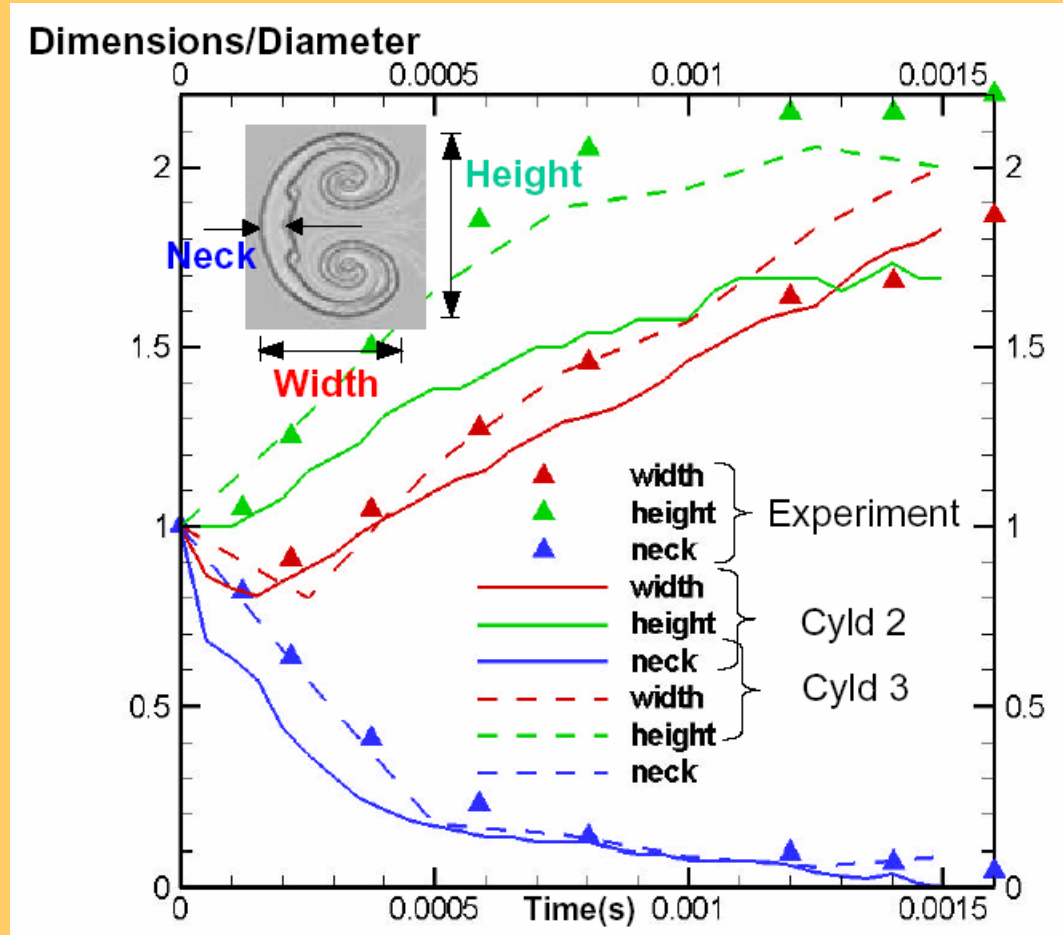


Comparing with Jacobs' Experiment cont.

Experiment PLIF Cycle 3 Cylid 3

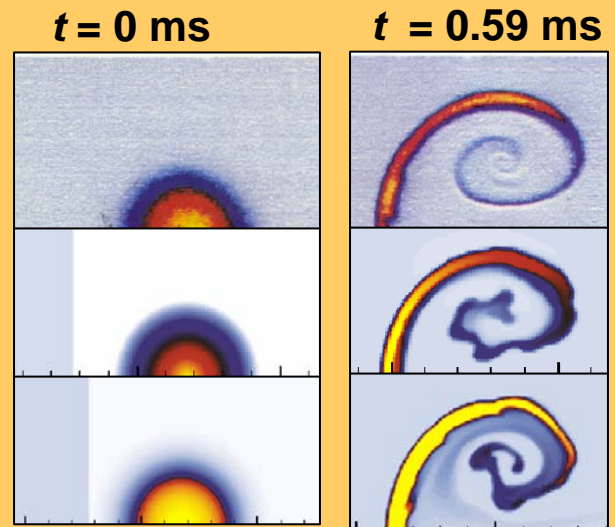


Variation of transition layer thickness: IC uncertainty



Sim I: Linear transition profile;

Sim II: Error function profile, with initial SF6 concentration 60%



Integrated vorticity space (x)- time (t) diagram (Hawley & Zabusky:PRL 1989)

